



Calhoun: The NPS Institutional Archive

Theses and Dissertations

Thesis Collection

1948-05

Method of application of moment distribution to solution of arched bents

Hansen, Bernard L.

Rensselaer Polytechnic Institute

<http://hdl.handle.net/10945/6493>



Calhoun is a project of the Dudley Knox Library at NPS, furthering the precepts and goals of open government and government transparency. All information contained herein has been approved for release by the NPS Public Affairs Officer.

**Dudley Knox Library / Naval Postgraduate School
411 Dyer Road / 1 University Circle
Monterey, California USA 93943**

<http://www.nps.edu/library>

METHOD OF APPLICATION OF MOMENT
DISTRIBUTION TO SOLUTION OF
ARCHED BENTS

BERNARD L. HANSEN

RICHARD O. JONES



1
1

Postgraduate School.
U. S. Naval Academy,
Annapolis, Md.

METHOD OF APPLICATION OF MOMENT DISTRIBUTION
TO SOLUTION OF ARCHED BENTS

By

Bernard L. Hansen
and
Richard O. Jones

Submitted to the Faculty of
Rensselaer Polytechnic Institute
in partial fulfillment of the requirements
for the degree of Master of Civil Engineering

Troy, New York

May 1948

We wish to express our appreciation to
Professor J. S. Kinney for his advice
and guidance in the compilation of this
thesis.

TABLE OF CONTENTS

I.	INTRODUCTION	PAGE
(a)	Object	1
(b)	Abstract	2
(c)	Definitions	3
(d)	Sign Convention	5
II.	PROCEDURE	
(a)	Conjugate Structure Method ..	6
(b)	Determination of FEM & Thrust	9
(c)	Determination of COF	15
(d)	Determination of absolute stiffness	17
(e)	Effect of spread	18
III	RESULTS	
(a)	Conclusions	19
(b)	Influence lines for FEM, thrust, and shear for parabolic arches with rise over span ratios from 0.04 to 0.40.....	20
(c)	Curves, FEM vs Rise/Span...	25
(d)	Curves, V_L vs Rise/Span...	27
IV.	Application OF RESULTS, to solution of an arched bent by moment distrib.	28
V.	APPENDIX	
(A)	Method A-determination of arc lengths	37
(B)	Method B-determination of arc lengths and tabulation of arc lengths-all spans....	38

	PAGE
(C) Determination of Neutral Point ...	41
(D) Calculations of influence lines for thrust, and shear, and FEM....	44
(E) Carry over factor, the effect of spread, and absolute stiffness calculations.....	53
VI BIBLIOGRAPHY.....	58

I N T R O D U C T I O N

OBJECT

This work is an extension of the investigation by Cain¹ into the practicability of applying the principles of moment distribution to the solution of arched bents. Cain developed curves for fixed end moment, carry-over-factor, and thrust in arches of 25' span for various rise/span ratios.

We have determined that a direct linear relationship exists between these factors for any span length and have provided curves for a parabolic arch based on a 1' span from which the values for any span can be readily computed.

¹CAIN, M. G., Application of Moment Distribution to Arched Bents, Thesis, RPI February, 1947

ABSTRACT

A method of applying the principles of moment distribution to parabolic arched bents is presented here-in.

The primary object has been to plot influence lines for the fixed-end moments, the horizontal reactions, and the vertical reactions at the ends of fixed-arches of any span length and rise/span ratio. These curves were computed by the conjugate structure method of analysis developed by Joseph S. Kinney, D.C.E.; Associate Professor of Structural Engineering, Rensselaer Polytechnic Institute, Troy, N.Y.

The authors claim no originality in regard to the concept of applying moment distribution to arches since it was recognized by Hardy Cross when he first presented his method, and is the basis of a similar work presented by Cain for one span length only.

DEFINITIONS

Carry-Over Factor....The relationship between the moment at the fixed end of a member and the moment applied at the opposite end which is hinged. Numerically, it is equal to the induced moment at the fixed end when a unit moment is applied at the hinged end.

Absolute Stiffness... The moment required to produce unit rotation of the hinged end of a member, whose far end is fixed.

Fixed End Moment..... The moment which would exist at the ends of a member if its ends were rigidly fixed against rotation.

Distribution Factor.. The ratio of the stiffness of one of the members at a joint to the sum of the stiffnesses of all the members at the joint.

Δ_H Total horizontal deflection of a point

Δ_V Total vertical deflection of a point

θ Total angular rotation of a point

δ_{VH} Vertical deflection of a point due to the application of a unit horizontal force at the point.

δ_{VV} Vertical deflection of a point due to the application of a unit vertical force.

δ_{VM} Vertical deflection of a point due to the application of a unit moment.

δ_{HH} Horizontal deflection of a point due to the application of a unit horizontal force.

δ_{HV} Horizontal deflection of a point due to the application of a unit vertical force.

δ_{HM} Horizontal deflection of a point due to the application of a unit moment.

$\delta_{\theta H}$	Angular rotation of the member at a point due to the application of a unit horizontal force at the point.
$\delta_{\theta V}$	Angular rotation of the member at a point due to the application of a unit vertical force at the point.
$\delta_{\theta M}$	Angular rotation of the member at a point due to the application of a unit moment at the point.
S	Effective span of the arch in feet.
\widehat{L}	Total arc length of the arch in feet.
ds	Arc length of a segment of the arch in feet.
y	Distance from the crown of the arch to the neutral point in feet.
x	Horizontal distance from a reference point, usually the left end of the arch, to the center of the arch segment in feet.
x'	The horizontal distance from the neutral point to the point on the arch where the M/EI loading on an arch segment is assumed to be concentrated, measured in feet.
y	The vertical distance from a reference point, usually the left springing, to the center of the arch segment in feet.
y'	The vertical distance from the neutral point to the point on the arch where the M/EI loading on an arch segment is assumed to be concentrated, measured in feet.
Rise ...	The vertical distance from the springing to crown in feet.

SIGN CONVENTION

The carry over factor is considered positive when the applied moment at the hinged end produces tension in the top of the member and the induced moment at the fixed end produces tension in the bottom of the member.

In the tabulated results for the fixed end moments, a positive moment is that moment which causes tension in the top fibers of the member.

In using the moment distribution method of solution, a positive moment is that internal moment which tends to rotate the joint in a clockwise direction.

PROCEDURE

As pointed out by Hardy Cross in the original presentation of his method of solving indeterminate rigid frames by "moment distribution", "It makes no difference whether the members are of constant or of varying cross-section, curved or straight, provided the constants: (a) stiffness at each end, (b) fixed-end moments at each end, and (c) carry-over factor at each end, are known or can be determined."

Since many structures have been constructed with curved top members on rigid frames, and since the usual method of solution for the stresses in such members is long and tedious, it appears that any information that will permit the engineer to apply the simple method of moment distribution to their solution will be of considerable value. We feel that there is much work to be done along this line and we have simply gone one step further than Cain in providing data which permits application of the method to spans of any length and for many rise/span ratios. Future investigation to determine these constants for any practical rise/span ratio and for members of varying cross-section is desirable.

In our work we have dealt with a parabolic arch of constant moment of inertia, and rise/span values of 0.04, 0.08, 0.20, 0.30, 0.40. Solutions were obtained for 20 and 30 foot spans, and it was determined that all the required constants varied in some direct relationship with the span of the arch. Consequently, all tabulated results and curves are based upon a unit span of 1'-0".

The values obtained were found by the use of "The Conjugate Structure Method " of analysis developed by Joseph Sterling Kinney, D. C. E.; Associate Professor of Structural Engineering, Rensselaer Polytechnic Institute, Troy, N. Y. Its method of application is described below.

"The conjugate structure, for a given real structure, is identical, in the lengths of its members and their relative position, to the real structure. It is considered to be positioned, however, so that it is located in a horizontal plane with the loads of the M/EI diagrams acting in a vertical direction.

The end of the conjugate structure, which corresponds to the end of the real structure which is permitted to deflect, if such an end exists, is always considered to be fixed in the conjugate structure.

The moment at any point on the conjugate structure (such moment being that caused by the M/EI diagram

loading) is the deflection of the corresponding point on the real structure in a direction perpendicular to the lever arms causing the moment in the conjugate structure.

The shear at any section of the conjugate structure is the rotation of the corresponding section of the real structure.

Sign Convention (for the conjugate structure)

Any moment in the real frame which causes compression of the inside surface thereof is considered to result in an M/EI diagram acting down on the conjugate structure.

If the moment at any section of the conjugate structure results in tension on the top fiber, the vertical deflection at the corresponding section of the real structure is down and the horizontal deflection is such as to shorten the horizontal projection of the distance between given points on the real structure.

If a section be passed through any point on the conjugate structure and if the portion of the conjugate structure thereby cut free tends to move down, then the angular rotation of the corresponding point on the real structure is counterclockwise."¹

1. Course Notes of Professor J. S. Kinney, 9/25/47

FIXED END MOMENT AND THRUST DETERMINATION:

In the determination of the fixed end moments, we decided to use a combination of the Neutral Point Method and the Conjugate Structure Method, thereby eliminating the need of solving simultaneous equations.

The arch is cut at the left reaction, cantilevering out from the right end, and a weightless bracket of infinite cross-section rigidly connects the cut end with the neutral point.

The arch is divided into ten equal horizontal projections, and the length of each arc segment determined as shown in appendix "A". It will be noted that two methods were used in the calculation of these arc lengths. Method "A" was used for the flat arches up to and including 0.20, and consists of using the first few terms of the infinite series which expresses the arc length of a parabola. The expressions used and a sample of the calculations appear in appendix "A". Method B was used for the higher values of rise/span since it was found that too many terms of the infinite series were required to accurately determine the arc lengths. Method "B" uses the exact logarithmic expression for arc length and sample calculations are to be found in Appendix "B".

The neutral point, the centroid of the conjugate structure, was computed by taking moments of the elastic areas about an axis through the crown as indicated in appendix "C". In the tables for computation of the location of the centroids it will be noted that the values of Ad^2 were determined and recorded as I_{yy} and I_{xx} . In normal calculations a designer would neglect the effects of the I_{cg} and the rotation of the axis for each segment which would be required for an accurate computation of the Moment of Inertia values. In checking our conjugate structure solution by the analagous column method, we found it was necessary to include the effect of the I_{cg} and the rotation of the axes in determining the moments of inertia in order to make the two methods agree. For slide rule accuracy it is sufficiently accurate to use the Ad^2 value as the moment of Inertia, but in the calculator accuracy as contained herein, the error was very appreciable.

The next step in our calculations was to determine the deflections of the neutral point due to the application at the neutral point of unit horizontal and vertical loads, and a unit moment. These were recorded as δ_{vv} , δ_{HH} , & $\delta_{\phi M}$ in which

δ_{VV} was the moment about the $x'-x'$ axis when the conjugate structure was loaded with the M/EI diagram for a unit vertical load at the neutral point; δ_{HH} was the moment about the $x'-x'$ axis, when the conjugate structure was loaded with the M/EI diagram for a unit horizontal load at the neutral point; and $\delta_{\theta M}$ was the shear at the neutral point when the conjugate structure was loaded with the M/EI diagram for a unit moment applied at the neutral point. Since the value of M/EI is a constant across the arch when a unit moment is applied at the neutral point, $\delta_{\theta M}$ was equal to the total arc length.

By definition of the neutral point, a horizontal force at the neutral point causes only horizontal deflection of the neutral point, a unit moment causes only rotation, and a unit vertical force causes only vertical deflection. All deflections being in the same direction as the applied unit force or moment. See Figure 1.

Next, unit vertical loads were placed at each of the 1/10 points across the span, and the deflections of the neutral point determined by means of the conjugate structure. These deflections were recorded as $\Delta_V, \Delta_H, \text{ \& } \theta$

It should be pointed out at this time that in the solution of our problems, we divided the M/EI diagram loading on a given arc segment into a triangle and a rectangle, and considered the total weight of the rectangle to be concentrated at the middle of the horizontal projection and located on the arc itself. The total weight of the triangle was considered to be concentrated at the $1/3$ point of the horizontal projection from the base of the triangle, and located on the arc itself at that horizontal projection. See figure 2.

From the theories of the neutral point method, the following relationships hold true:

$$H_o = - \frac{\Delta H}{\delta_{HH}} = H_L$$

$$V_o = - \frac{\Delta v}{\delta_{vv}} = V_L$$

$$M_o = - \frac{\theta}{\delta_{\theta M}}$$

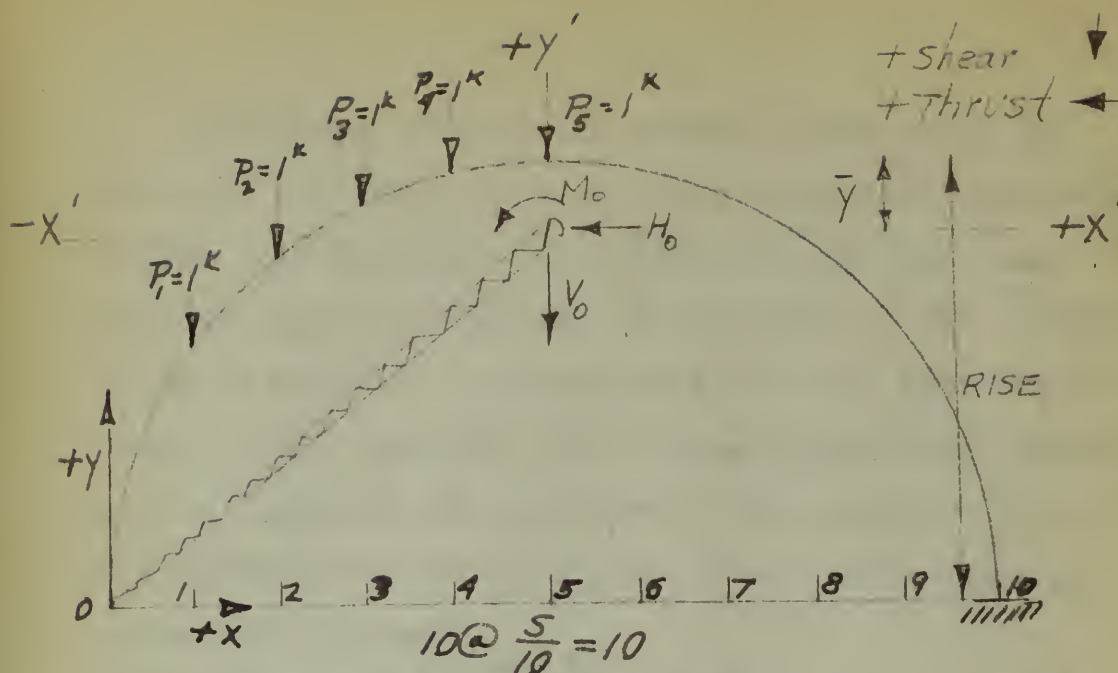
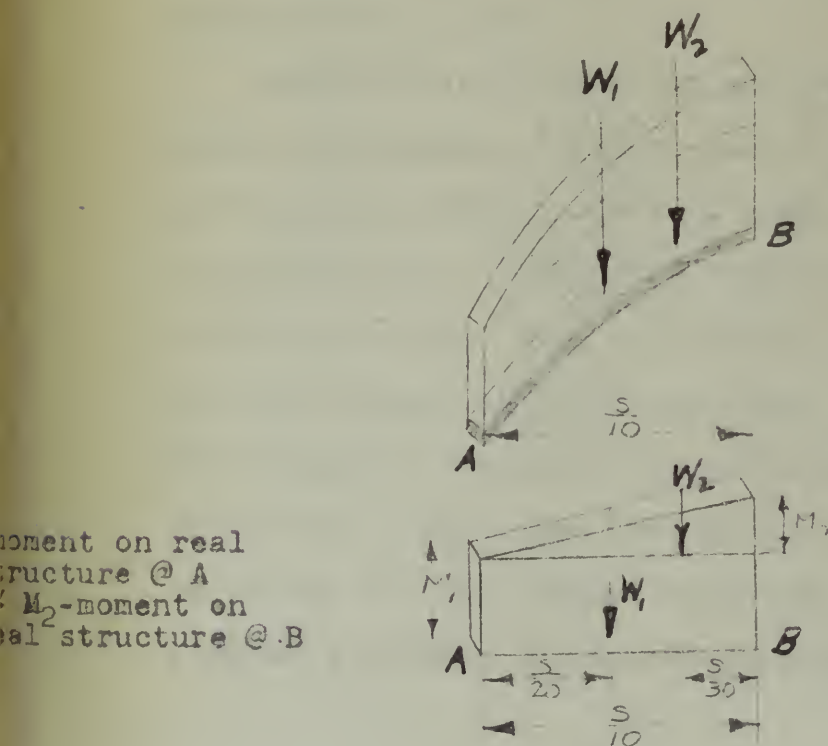


FIGURE 1 Parabolic Arch

Redundant reactions applied at neutral point and vertical loads applied at 1/10 th span points.



M/EI diagram loading on an arc--divided into rectangle and triangle sections.

Horizontal Projection of M/EI diagram loading on the arc.

W_1 -elastic wt. of rect.

W_2 -elastic wt. of triangle

ds -actual arc length

FIGURE 2

The Shear, thrust, and moment at any point on the real structure can readily be determined by drawing a free body diagram of the arch with the right end rigidly fixed and the left end attached to the neutral point by means of the weightless bracket, applying the loads on the structure and the reactions at the neutral point. Applying the reactions at the neutral point in the directions shown in Figure 1, and using the sign convention thereof,

$$M_L = M_0 + H_0 (\text{RISE} - \bar{y}) - V_0 \frac{S}{2}$$

$$M_R = M_0 + H_0 (\text{RISE} - \bar{y}) + V_0 \frac{S}{2} + P (S - x)$$

with a positive moment indicating compression on the inside of the arch.

Sample calculations for the determination of the fixed end moments, vertical shear, and horizontal reactions for various locations of unit loads on the arch are shown in Appendix "D". Tabulated results for the various rise/span ratios, with all values expressed in terms of the span length in feet, are shown on Page 52. The influence lines for the fixed end moments, the horizontal thrusts and the vertical shears at the springing for the various rise/span ratios are shown in Figures 3 through 10 inclusive, pages 20-24.

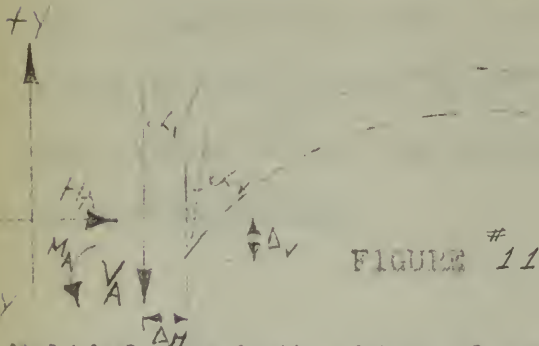
DETERMINATION OF CARRY-OVER FACTOR

When an arch of constant moment of inertia is cut at the left springing, and redundant forces applied as shown in the figure below, deflection of the left springing will be as follows:

$$(1) \Delta_V = M_A \delta_{VM} + V_A \delta_{VV} + H_A \delta_{VH}$$

$$(2) \Delta_H = M_A \delta_{HM} + V_A \delta_{HV} + H_A \delta_{HH}$$

$$(3) \theta = M_A \delta_{\theta M} + V_A \delta_{\theta V} + H_A \delta_{\theta H}$$



Multiplying both sides of equations 1, 2, & 3 by EI results in:

$$(4) EI \Delta_V = M_A (EI \delta_{VM}) + V_A (EI \delta_{VV}) + H_A (EI \delta_{VH})$$

$$(5) EI \Delta_H = M_A (EI \delta_{HM}) + V_A (EI \delta_{HV}) + H_A (EI \delta_{HH})$$

$$(6) EI \theta = M_A (EI \delta_{\theta M}) + V_A (EI \delta_{\theta V}) + H_A (EI \delta_{\theta H})$$

From the theories of virtual work and Maxwells Law of reciprocal deflections:

$$EI \delta_{VM} = EI \delta_{\theta V} = \int \epsilon x ds = \widehat{L} \cdot \frac{s}{2}$$

$$EI \delta_{HM} = EI \delta_{\theta H} = \int \epsilon y ds = \widehat{L} \cdot (\text{RISE} - \bar{y})$$

$$EI \delta_{\theta M} = \int \epsilon ds = \widehat{L}$$

$$EI \delta_{VH} = EI \delta_{HV} = \int \epsilon xy ds = \widehat{L} \cdot \frac{s}{2} (\text{RISE} - \bar{y})$$

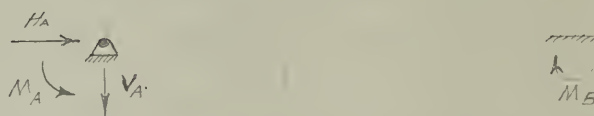
$$EI \delta_{VV} = \int \epsilon x^2 ds$$

$$EI \delta_{HH} = \int \epsilon y^2 ds$$

Since the carry-over factor is numerically equal to the moment induced at the fixed end of a member when a unit moment produces rotation at the hinged end, there will be no vertical or horizontal deflections.

Consequently equations (4) and (5) are equal to zero.

Solving equations (4) & (5) simultaneously will give us the forces induced at the hinged end when a unit moment is applied. The carryover factor will be found by setting up the free body diagram indicated in the sketch below, and solving for the induced moment at the fixed end.



When the induced moment at the fixed end is such that it causes tension on the same side of the member at the fixed end as the applied moment causes on the hinged end, the carry-over factor is considered negative.

It is evident that the application of a moment at the hinged end will introduce a horizontal thrust. This thrust must be taken into account in the solving of an arched bent by moment distribution since every time a

joint is rotated and the moments balanced out at the joint, a moment will be applied to the arch. Such applied moment will change the horizontal restraining force at the ends of the arch.

Sample calculations for the determination of the carry-over factor and horizontal restraining forces will be found on page 3 of Appendix "E". The carry over factors and horizontal restraining forces are tabulated in Appendix "E", in terms of the length of span, and rise/span ratios.

ABSOLUTE STIFFNESS

The absolute stiffness is defined as the moment required to produce unit rotation of the hinged end of a member, whose far end is rigidly fixed. Thus, there will be no horizontal or vertical deflection at the hinged end, and equations (4) & (5) are again equal to zero.

Solving these equations simultaneously will give us the forces induced at the hinged end in terms of the applied moment. Substituting these values in equation (6) will give us $EI\theta$ expressed in terms of the applied moment. Setting $\theta = 1$ for unit rotation, and solving for M_A , we will get the absolute stiffness of the arch. After solving several span lengths with the same rise/span ratio, it was determined that the absolute stiffness varied inversely as the span. Consequently the tabulated

values of the absolute stiffness for each rise/span ratio are in terms of EI/S . Sample calculations for the determination of the absolute stiffness are shown on Page 3 of Appendix "E".

SPREAD

It has been shown in the calculations for the fixed end moments that every load placed on the arch causes a horizontal thrust at the end. This thrust will cause spread in the arched bent and must be taken into consideration when solving the arched bent by moment distribution.

If a spread of Δ_H feet occurs while there is no vertical deflection or rotation of the joint, reactions will be induced at the ends of the arch. These reactions will be determined by setting equations (4) & (6) equal to zero and solving equations (4), (5), & (6) simultaneously for H_A , V_A , and M_A in terms of EI . After solving several spans for the same rise/span ratio, it was determined that V_A was equal to zero in all cases and that M_A varied inversely as the square of the span, and H_A varied inversely as the cube of the span. Therefore the induced moment was tabulated in terms of $\frac{EI \Delta_H}{S^2}$ and the thrust in terms of $\frac{EI \Delta_H}{S^3}$ for all values of Rise/Span. Results are tabulated on page 57 and sample calculations are shown on page 56.

CONCLUSIONS

In this paper, we have presented curves and tabulated data which will permit an engineer to solve any parabolic arched bent by moment distribution regardless of loading, provided the arched member is of constant moment of inertia and has rise/span ratios between 0.04 and 0.40.

When similar curves and data are determined for arched members with varying moment of inertia across the arch, all parabolic arched bents may be solved with great saving in time and energy by the method of moment distribution.

As a result of this investigation, it was found that for a given rise/span ratio; the following relationships are true:

1. The shear and thrust are constant for all spans.
2. The F.E.M. is directly proportional to the span,
i.e. F.E.M. for 20' span = 20 x FEM for 1' span.
3. The carry over factor is constant for all spans.
4. The absolute stiffness varies inversely as
the span.

CURVES FOR 1-FOOT SPAN

$$\frac{\text{RISE}}{\text{SPAN}} = 0.04$$

$$\text{C.O.F.} = 0.3505$$

FT-KIPS $\times 10$

0.5

0.0

.0155

.0165

FIXED END MOMENTS

$$\text{CONC. LOAD } M_L^F = 0.1 \cdot f \cdot S \cdot W$$

$$\text{UNIF. LOAD } M_L^F = 0.001 S^2 \cdot W$$

$$S = \text{SPAN - FT}$$

f = factor from CURVES

W = CONC. LOAD - KIPS

W = UNIF. LOAD KIPS/FT

KIPS

1.0

0.0

VERTICAL FORCE

V₀

KIPS

3.0

2.0

1.0

0.0

THRUST

CONC. LOAD = READ CURVE

$$\text{UNIF. LOAD} = 3.004 \times W \times S$$

THRUST

H₀

0 1 2 3 4 5 6 7 8 9 10

LOAD POSITIONS

CURVES FOR ONE-FOOT SPAN

RISE
SPAN = 0.06

C.O.F = 0.3497

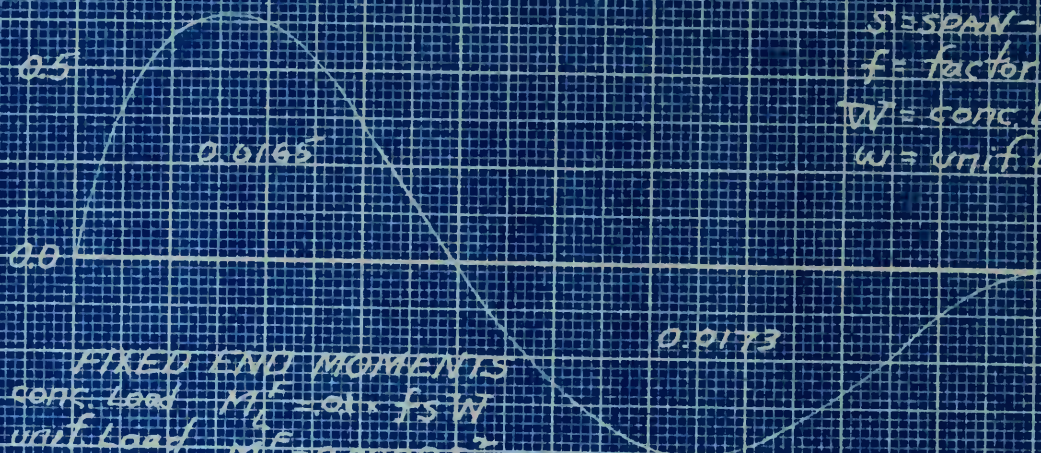
S = SPAN - FT

f = factor from CURVES

W = CONC. Load - KIPS

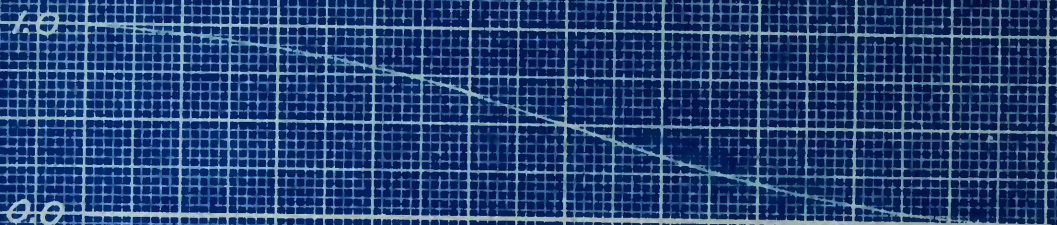
w = unif. Load - KIPS/FT

FT. KIPS x 10



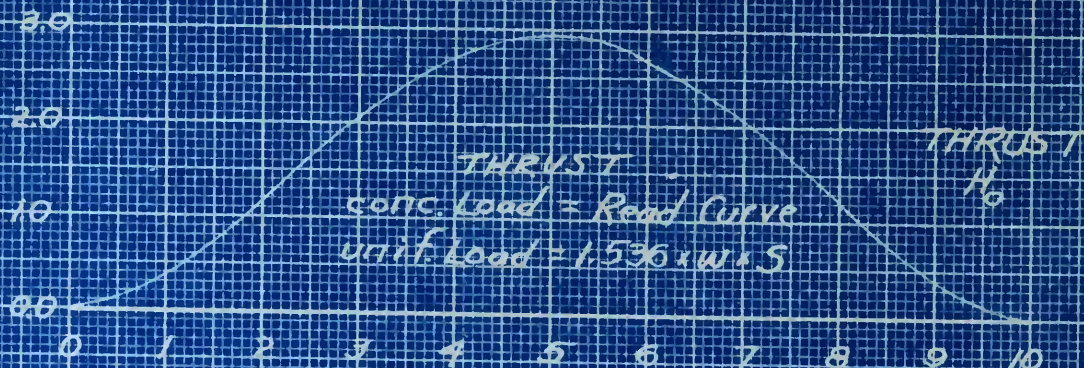
FIXED END MOMENTS
 CONC. Load $M_L^F = 0.01 \times f \times S \times W$
 unif. Load $M_L^F = 0.00085 \times W$

KIPS



VERTICAL FORCE
 $\frac{1}{6}$

KIPS



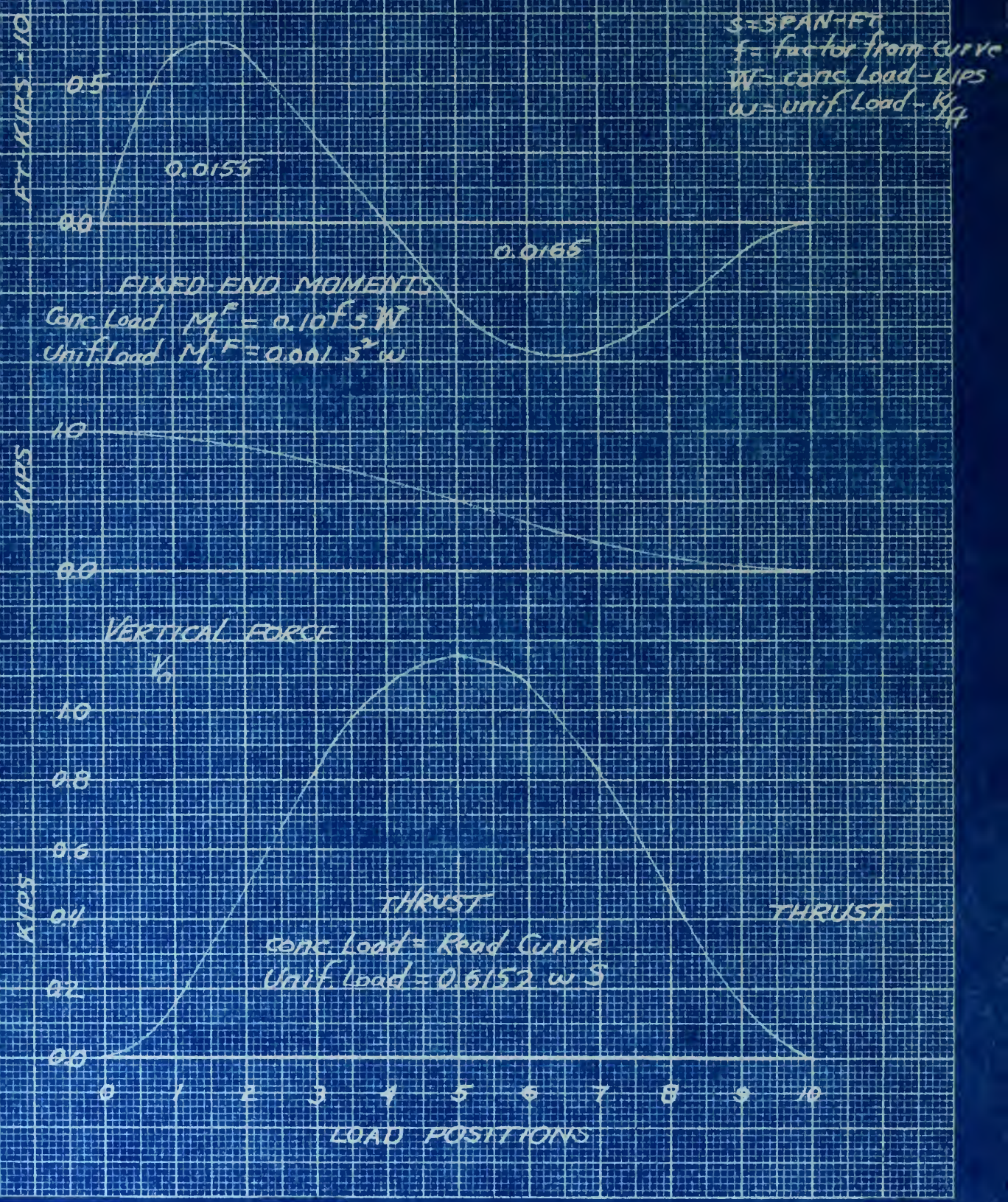
THRUST
 CONC. Load = Read Curve
 unif. Load = $1.536 \times W \times S$

THRUST
 H_0

LOAD POSITIONS

CURVES FOR ONE-FOOT SPAN

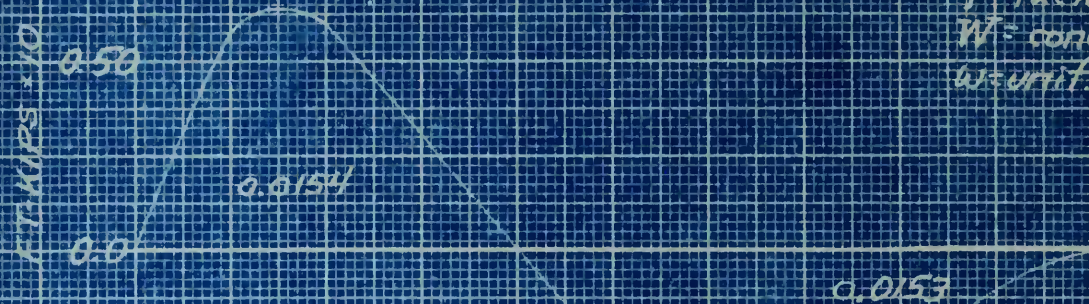
$\frac{\text{RISE}}{\text{SPAN}} = 0.20$ C.O.F. = 0.3481



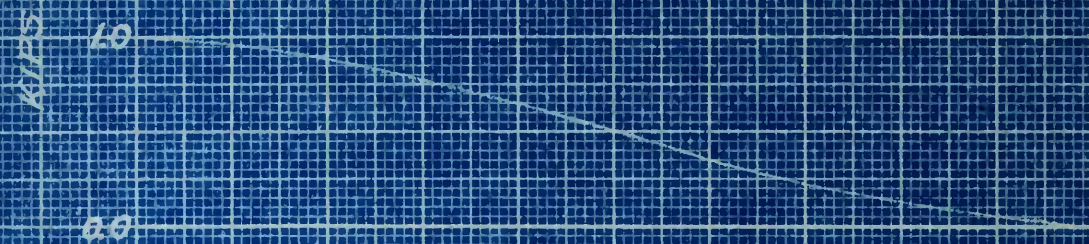
CURVES FOR ONE-FOOT SPAN

$$\frac{\text{RISE}}{\text{SPAN}} = 0.30 \quad \text{C.D.F.} = 0.3384$$

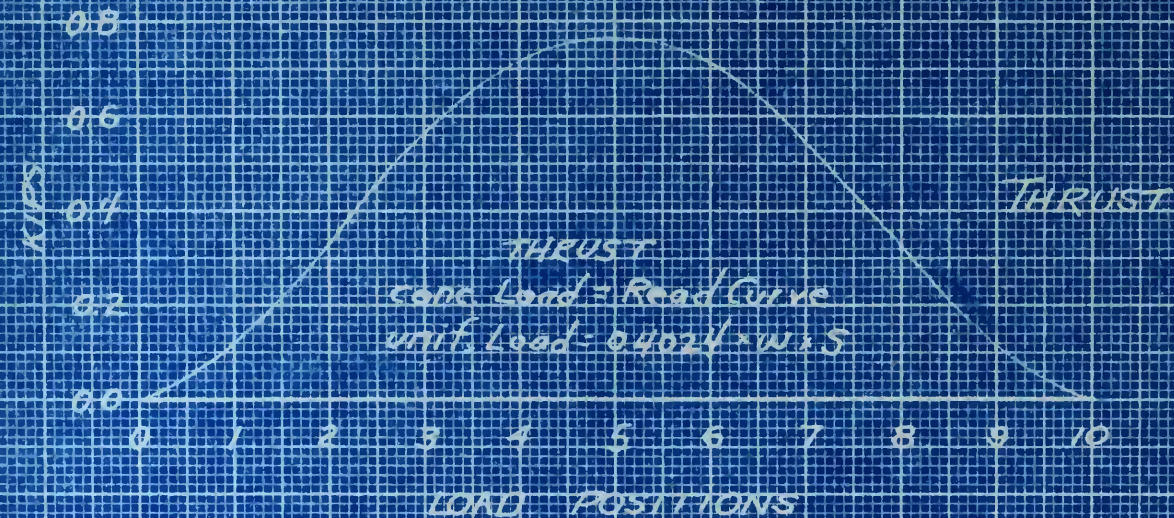
$S = \text{SPAN} - \text{FT}$
 $f = \text{factor from curves}$
 $W = \text{conc. Load - KIPS}$
 $w = \text{unif. Load - KIPS/ft}$



FIXED END MOMENTS
 conc. Load $M_1^F = 0.154 f \cdot S \cdot W$
 unif. Load $M_1^F = 0.0001 \cdot S^2 \cdot w$



VERTICAL FORCE
 $\frac{1}{2}$



THRUST
 conc. Load = Read Curve
 unif. Load = $0.4024 \cdot w \cdot S$

CURVES FOR ONE-FOOT SPAN

$$\frac{\text{RISE}}{\text{SPAN}} = 0.40$$

$$\text{C.O.F.} = 0.3362$$

$S = \text{SPAN} - \text{FT}$

$f = \text{factor from curves}$

$W = \text{CONC. LOAD} - \text{KIPS}$

$w = \text{UNIF. LOAD} - \text{K/FT}$

FT. KIPS $\times 10$

0.5

0.0

0.0149

0.0152

FIXED END MOMENTS

CONC. LOAD $M_1^F = 0.1 f S W$

UNIF. LOAD $M_1^F = 0.0003 S^2 w$

KIPS

10

0.0

VERTICAL FORCE

V_6

0.4

0.3

0.2

0.1

0.0

THRUST

CONC. LOAD - READ CURVE

UNIF. LOAD - $0.3044 \times W \times S$

THRUST

KIPS

0

1

2

3

4

5

6

7

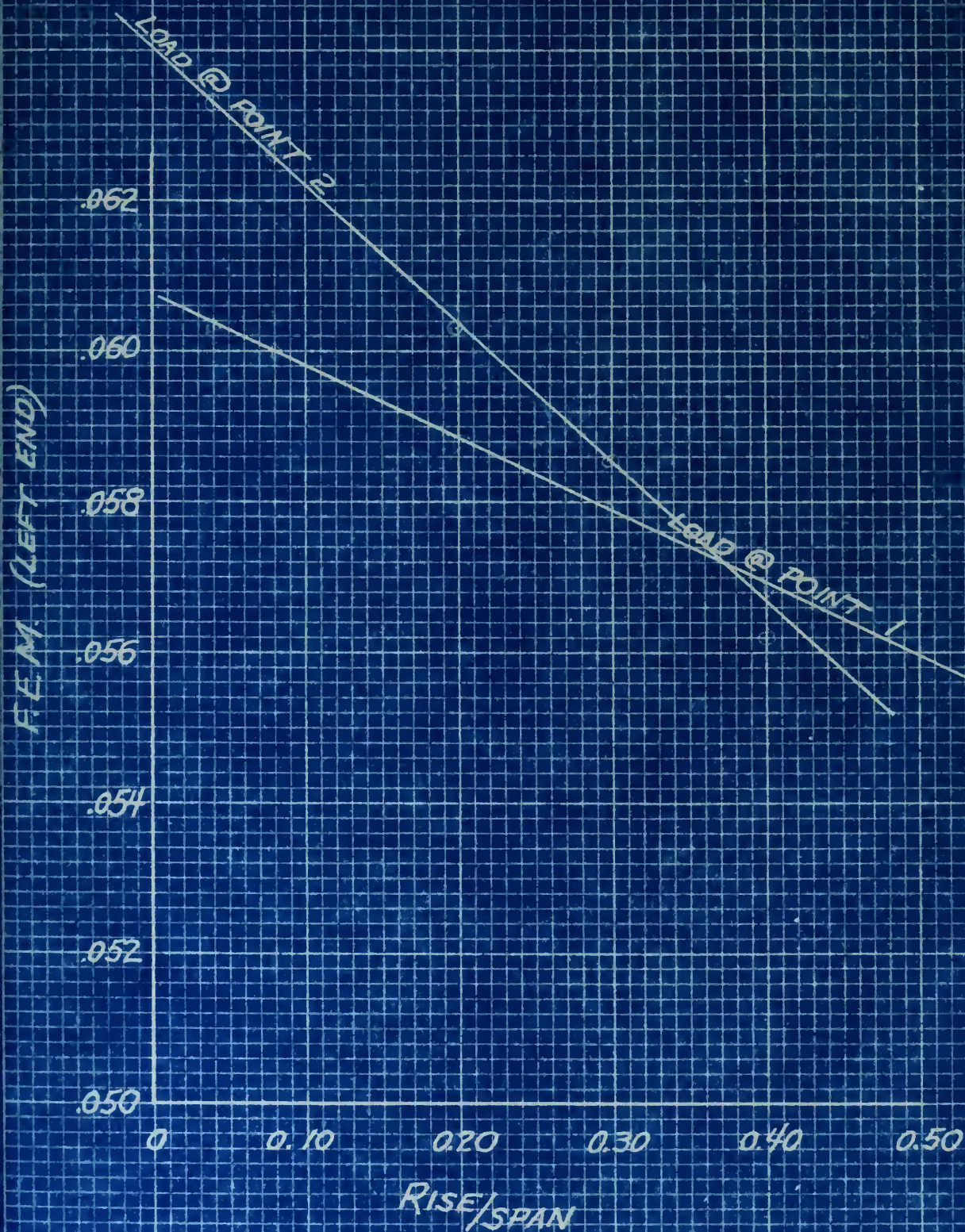
8

9

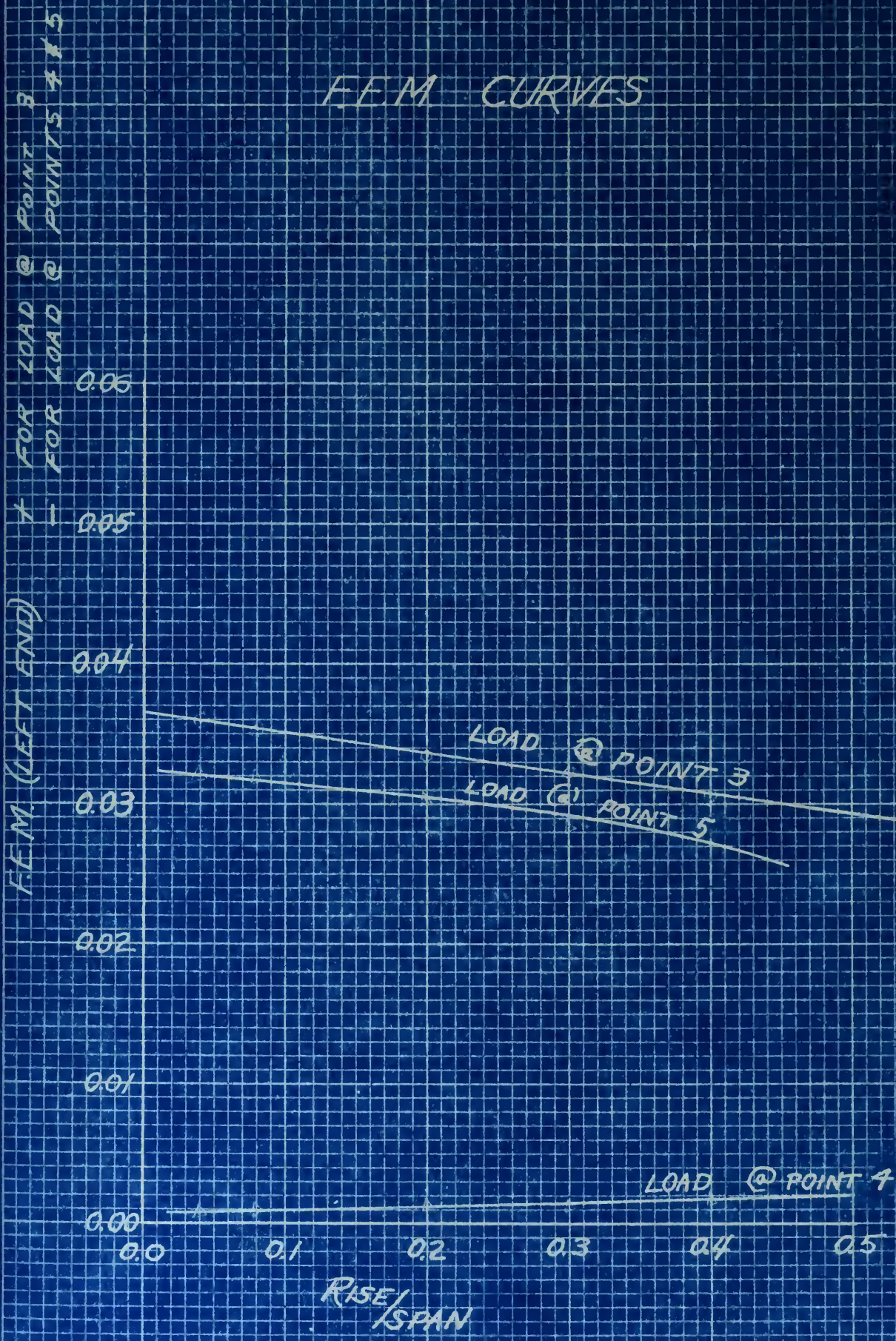
10

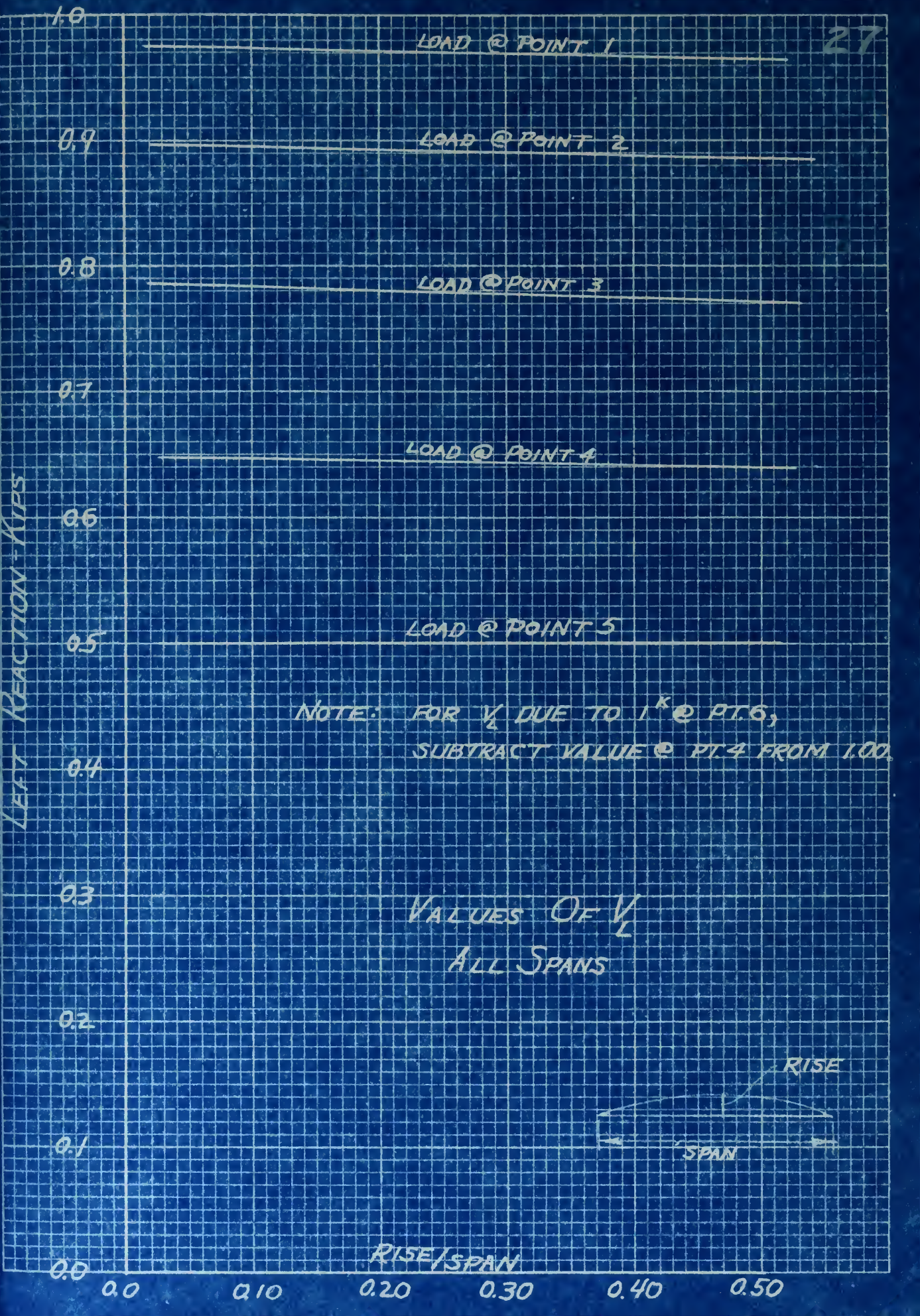
LOAD POSITIONS

FEM CURVES

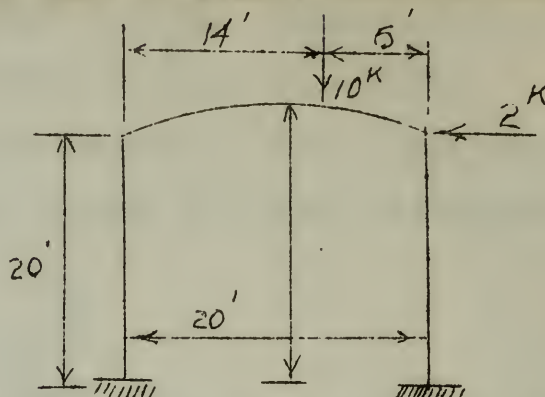


FEM CURVES





APPLICATION OF RESULTS TO THE SOLUTION OF AN ARCHED BENT BY MOMENT DISTRIBUTION.



GIVEN: An arched bent having a span of 20'-0" and a parabolic rise of 4'-0" as shown in the above sketch. All members have the same moment of Inertia. Determine the reactions at the supports and the moments at B and C, caused by the given loads on the bent.

From the curves on page 22 , the following values for the fixed end arch conditions can be determined:

$$M_{BC}^R = \frac{-(0.460) 20}{10} = -9.20'k$$

$$M_{CB}^F = \frac{-(0.334) 20}{10} = -6.68$$

$$H_B = 8.30 \text{ kips} \longrightarrow$$

$$H_C = 8.30 \text{ kips} \longleftarrow$$

with a positive moment indicating tension on the top of the member. Since a positive moment as used in moment distribution is that moment which tends to rotate the joint clockwise,

$$M_{BC}^F = -9.20 \text{ ft.kips} \quad M_{CB}^F = -6.68 \text{ ft. kips}$$

The absolute stiffness of the arch, from page 57 , is equal to $7.8855EI/S$ which equals $0.3943 EI$. The carry-over factor is -0.343 .

The absolute stiffness of the prismatic column members is equal to $\frac{4EI}{L}$ which equals $0.200EI$ and the carry over factor is equal to $\frac{1}{2} 0.500$.

Assuming rotation of the joints only, solve for the moments at all joints by moment distribution.

JOINT	A	B		BALANCING MOMENTS	C		D
MEMB.	AB	BA	BC		CB	CD	DC
K	.200	.200	.3943		.3943	.200	.200
$\frac{K}{\sum k}$	-	.337	.663		.663	.337	-
COF	$\frac{1}{2} .5$			$-.343$		$\frac{1}{2} .50$	
FEM							
	$\frac{1}{2} 1.55$	$\frac{1}{2} 3.10$	$\frac{1}{2} 6.10$		$\frac{1}{2} 5.80$	$\frac{1}{2} 2.97$	$\frac{1}{2} 1.48$
	$\frac{1}{2} 0.32$	$\frac{1}{2} 0.64$	$\frac{1}{2} 1.34$		$\frac{1}{2} 0.30$	$\frac{1}{2} 0.15$	$\frac{1}{2} 0.08$
M_R	$\frac{1}{2} 1.89$	$\frac{1}{2} 3.77$	$\frac{1}{2} -3.77$		$\frac{1}{2} -3.13$	$\frac{1}{2} 3.13$	$\frac{1}{2} 1.56$
Bal. Moms.				$\frac{1}{2} 7.51$	$\frac{1}{2} 6.12$		

The chart shown differs from the usual moment distribution chart in that a record is kept of the balancing moments taken by the arch when the joint is permitted to rotate. This balancing moment changes the horizontal restraining force at the ends of the curved member by an amount dependent on the properties of the arch in question.

The following table shows the results of the experiments conducted on the 15th of May 1881. The experiments were conducted on the 15th of May 1881. The results of the experiments are as follows:

Time	Temperature	Pressure	Volume	Weight
1.00	20.0	760.0	1.00	1.00
1.10	20.5	760.5	1.05	1.05
1.20	21.0	761.0	1.10	1.10
1.30	21.5	761.5	1.15	1.15
1.40	22.0	762.0	1.20	1.20
1.50	22.5	762.5	1.25	1.25
2.00	23.0	763.0	1.30	1.30
2.10	23.5	763.5	1.35	1.35
2.20	24.0	764.0	1.40	1.40
2.30	24.5	764.5	1.45	1.45
2.40	25.0	765.0	1.50	1.50
2.50	25.5	765.5	1.55	1.55
3.00	26.0	766.0	1.60	1.60
3.10	26.5	766.5	1.65	1.65
3.20	27.0	767.0	1.70	1.70
3.30	27.5	767.5	1.75	1.75
3.40	28.0	768.0	1.80	1.80
3.50	28.5	768.5	1.85	1.85
4.00	29.0	769.0	1.90	1.90
4.10	29.5	769.5	1.95	1.95
4.20	30.0	770.0	2.00	2.00
4.30	30.5	770.5	2.05	2.05
4.40	31.0	771.0	2.10	2.10
4.50	31.5	771.5	2.15	2.15
5.00	32.0	772.0	2.20	2.20
5.10	32.5	772.5	2.25	2.25
5.20	33.0	773.0	2.30	2.30
5.30	33.5	773.5	2.35	2.35
5.40	34.0	774.0	2.40	2.40
5.50	34.5	774.5	2.45	2.45
6.00	35.0	775.0	2.50	2.50
6.10	35.5	775.5	2.55	2.55
6.20	36.0	776.0	2.60	2.60
6.30	36.5	776.5	2.65	2.65
6.40	37.0	777.0	2.70	2.70
6.50	37.5	777.5	2.75	2.75
7.00	38.0	778.0	2.80	2.80
7.10	38.5	778.5	2.85	2.85
7.20	39.0	779.0	2.90	2.90
7.30	39.5	779.5	2.95	2.95
7.40	40.0	780.0	3.00	3.00
7.50	40.5	780.5	3.05	3.05
8.00	41.0	781.0	3.10	3.10
8.10	41.5	781.5	3.15	3.15
8.20	42.0	782.0	3.20	3.20
8.30	42.5	782.5	3.25	3.25
8.40	43.0	783.0	3.30	3.30
8.50	43.5	783.5	3.35	3.35
9.00	44.0	784.0	3.40	3.40
9.10	44.5	784.5	3.45	3.45
9.20	45.0	785.0	3.50	3.50
9.30	45.5	785.5	3.55	3.55
9.40	46.0	786.0	3.60	3.60
9.50	46.5	786.5	3.65	3.65
10.00	47.0	787.0	3.70	3.70
10.10	47.5	787.5	3.75	3.75
10.20	48.0	788.0	3.80	3.80
10.30	48.5	788.5	3.85	3.85
10.40	49.0	789.0	3.90	3.90
10.50	49.5	789.5	3.95	3.95
11.00	50.0	790.0	4.00	4.00
11.10	50.5	790.5	4.05	4.05
11.20	51.0	791.0	4.10	4.10
11.30	51.5	791.5	4.15	4.15
11.40	52.0	792.0	4.20	4.20
11.50	52.5	792.5	4.25	4.25
12.00	53.0	793.0	4.30	4.30
12.10	53.5	793.5	4.35	4.35
12.20	54.0	794.0	4.40	4.40
12.30	54.5	794.5	4.45	4.45
12.40	55.0	795.0	4.50	4.50
12.50	55.5	795.5	4.55	4.55
13.00	56.0	796.0	4.60	4.60
13.10	56.5	796.5	4.65	4.65
13.20	57.0	797.0	4.70	4.70
13.30	57.5	797.5	4.75	4.75
13.40	58.0	798.0	4.80	4.80
13.50	58.5	798.5	4.85	4.85
14.00	59.0	799.0	4.90	4.90
14.10	59.5	799.5	4.95	4.95
14.20	60.0	800.0	5.00	5.00
14.30	60.5	800.5	5.05	5.05
14.40	61.0	801.0	5.10	5.10
14.50	61.5	801.5	5.15	5.15
15.00	62.0	802.0	5.20	5.20
15.10	62.5	802.5	5.25	5.25
15.20	63.0	803.0	5.30	5.30
15.30	63.5	803.5	5.35	5.35
15.40	64.0	804.0	5.40	5.40
15.50	64.5	804.5	5.45	5.45
16.00	65.0	805.0	5.50	5.50
16.10	65.5	805.5	5.55	5.55
16.20	66.0	806.0	5.60	5.60
16.30	66.5	806.5	5.65	5.65
16.40	67.0	807.0	5.70	5.70
16.50	67.5	807.5	5.75	5.75
17.00	68.0	808.0	5.80	5.80
17.10	68.5	808.5	5.85	5.85
17.20	69.0	809.0	5.90	5.90
17.30	69.5	809.5	5.95	5.95
17.40	70.0	810.0	6.00	6.00
17.50	70.5	810.5	6.05	6.05
18.00	71.0	811.0	6.10	6.10
18.10	71.5	811.5	6.15	6.15
18.20	72.0	812.0	6.20	6.20
18.30	72.5	812.5	6.25	6.25
18.40	73.0	813.0	6.30	6.30
18.50	73.5	813.5	6.35	6.35
19.00	74.0	814.0	6.40	6.40
19.10	74.5	814.5	6.45	6.45
19.20	75.0	815.0	6.50	6.50
19.30	75.5	815.5	6.55	6.55
19.40	76.0	816.0	6.60	6.60
19.50	76.5	816.5	6.65	6.65
20.00	77.0	817.0	6.70	6.70
20.10	77.5	817.5	6.75	6.75
20.20	78.0	818.0	6.80	6.80
20.30	78.5	818.5	6.85	6.85
20.40	79.0	819.0	6.90	6.90
20.50	79.5	819.5	6.95	6.95
21.00	80.0	820.0	7.00	7.00
21.10	80.5	820.5	7.05	7.05
21.20	81.0	821.0	7.10	7.10
21.30	81.5	821.5	7.15	7.15
21.40	82.0	822.0	7.20	7.20
21.50	82.5	822.5	7.25	7.25
22.00	83.0	823.0	7.30	7.30
22.10	83.5	823.5	7.35	7.35
22.20	84.0	824.0	7.40	7.40
22.30	84.5	824.5	7.45	7.45
22.40	85.0	825.0	7.50	7.50
22.50	85.5	825.5	7.55	7.55
23.00	86.0	826.0	7.60	7.60
23.10	86.5	826.5	7.65	7.65
23.20	87.0	827.0	7.70	7.70
23.30	87.5	827.5	7.75	7.75
23.40	88.0	828.0	7.80	7.80
23.50	88.5	828.5	7.85	7.85
24.00	89.0	829.0	7.90	7.90
24.10	89.5	829.5	7.95	7.95
24.20	90.0	830.0	8.00	8.00
24.30	90.5	830.5	8.05	8.05
24.40	91.0	831.0	8.10	8.10
24.50	91.5	831.5	8.15	8.15
25.00	92.0	832.0	8.20	8.20
25.10	92.5	832.5	8.25	8.25
25.20	93.0	833.0	8.30	8.30
25.30	93.5	833.5	8.35	8.35
25.40	94.0	834.0	8.40	8.40
25.50	94.5	834.5	8.45	8.45
26.00	95.0	835.0	8.50	8.50
26.10	95.5	835.5	8.55	8.55
26.20	96.0	836.0	8.60	8.60
26.30	96.5	836.5	8.65	8.65
26.40	97.0	837.0	8.70	8.70
26.50	97.5	837.5	8.75	8.75
27.00	98.0	838.0	8.80	8.80
27.10	98.5	838.5	8.85	8.85
27.20	99.0	839.0	8.90	8.90
27.30	99.5	839.5	8.95	8.95
27.40	100.0	840.0	9.00	9.00
27.50	100.5	840.5	9.05	9.05
28.00	101.0	841.0	9.10	9.10
28.10	101.5	841.5	9.15	9.15
28.20	102.0	842.0	9.20	9.20
28.30	102.5	842.5	9.25	9.25
28.40	103.0	843.0	9.30	9.30
28.50	103.5	843.5	9.35	9.35
29.00	104.0	844.0	9.40	9.40
29.10	104.5	844.5	9.45	9.45
29.20	105.0	845.0	9.50	9.50
29.30	105.5	845.5	9.55	9.55
29.40	106.0	846.0	9.60	9.60
29.50	106.5	846.5	9.65	9.65
30.00	107.0	847.0	9.70	9.70
30.10	107.5	847.5	9.75	9.75
30.20	108.0	848.0	9.80	9.80
30.30	108.5	848.5	9.85	9.85
30.40	109.0	849.0	9.90	9.90
30.50	109.5	849.5	9.95	9.95
31.00	110.0	850.0	10.00	10.00
31.10	110.5	850.5	10.05	10.05
31.20	111.0	851.0	10.10	10.10
31.30	111.5	851.5	10.15	10.15
31.40	112.0	852.0	10.20	10.20
31.50	112.5	852.5	10.25	10.25
32.00	113.0	853.0	10.30	10.30
32.10	113.5	853.5	10.35	10.35
32.20	114.0	854.0	10.40	10.40
32.30	114.5	854.5	10.45	10.45
32.40	115.0	855.0	10.50	10.50
32.50	115.5	855.5	10.55	10.55
33.00	116.0	856.0	10.60	10.60
33.10	116.5	856.5	10.65	10.65
33.20	117.0	857.0	10.70	10.70
33.30	117.5	857.5	10.75	10.75
33.40	118.0	858.0	10.80	10.80
33.50	118.5	858.5	10.85	10.85
34.00	119.0	859.0	10.90	10.90
34.10	119.5	859.5	10.95	10.95
34.20	120.0	860.0	11.00	11.00
34.30	120.5	860.5	11.05	11.05
34.40	121.0	861.0	11.10	11.10
34.50	121.5	861.5	11.15	11.15
35.00	122.0	862.0	11.20	11.20
35.10	122.5	862.5	11.25	11.25
35.20	123.0	863.0	11.30	11.30
35.30	123.5	863.5	11.35	11.35
35.40	124.0	864.0	11.40	11.40
35.50	124.5	864.5	11.45	11.45
36.00	125.0	865.0	11.50	11.50
36.10	125.5	865.5	11.55	11.55
36.20	126.0	866.0	11.60	11.60
36.30	126.5	866.5	11.65	11.65
36.40	127.0	867.0	11.70	11.70
36.50	127.5	867.5	11.75	11.75
37.00	128.0	868.0	11.80	11.80
37.10	128.5	868.5	11.85	11.85
37.20	129.0	869.0	11.90	11.90
37.30	129.5	869.5	11.95	11.95
37.40	130.0	870.0	12.00	12.00
37.50	130.5	870.5	12.05	12.05
38.00	131.0	871.0	12.10	12.10
38.10	131.5	871.5	12.15	12.15
38.20	132.0	872.0	12.20	12.20
38.30	132.5	872.5	12.25	12.25
38.40	133.0	873.0	12.30	12.30
38.50	133.5	873.5	12.35	12.35
39.00	134.0	874.0		

For arched bents, the required horizontal restraining forces at the ends of the arch are made up of the following components:

- (a) That force necessary to restrain the horizontal thrust at the ends of the arch caused by applying a load on the arch or caused by spread in the arch.
- (b) That force necessary to restrain the horizontal thrust at the ends of the arch caused by the balancing moments.
- (c) That necessary to balance the shears in the column members.
- (d) That force necessary to prevent sidesway when a horizontal load is applied to the bent.

A positive balancing moment (using the sign convention of moment distribution) at the left springing causes equal horizontal restraining forces at the two ends of the arch acting in a direction away from each other. A positive balancing moment at the right springing causes equal horizontal restraining forces at the ends of the arch acting in a direction toward each other.

For the bent under consideration, the horizontal restraining forces at the ends of the arch caused by the balancing moments is equal to 4.2884 M/S which equals 0.21442 M. See page 57.

The Artificial joint restraints (AJR) required to prevent the original bent from deflecting due to sidesway or spread are equal to the horizontal restraining forces at the ends of the arch.

HRF caused by:	HRF _L $\xrightarrow{+}$	HRF _R
(a) Load on the arch	48.30	-8.30
(b) Balancing Moment at B (0.21442)(7.51)	-1.61	41.61
(c) Balancing Moment at C (0.21442)(6.12)	41.31	-1.31
(d) Shear in member AB $\frac{(1.89 \text{ } 3.77)}{20}$	40.28	
(e) Shear in member CD $\frac{(1.58 \text{ } 3.13)}{20}$		40.24
(f) 2 Kip load at C		42.00

$$\text{AJR}_L \xrightarrow{48.28} \text{AJR}_R \xleftarrow{-5.76}$$

If we let x equal artificial joint restraint resistant against spread and let y equal artificial joint restraint resisting sidesway, we have:

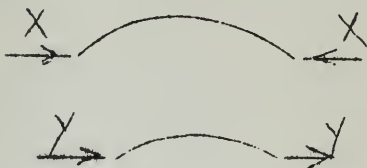
$$x + y = 8.28$$

$$x - y = 5.76$$

$$2x = 14.04$$

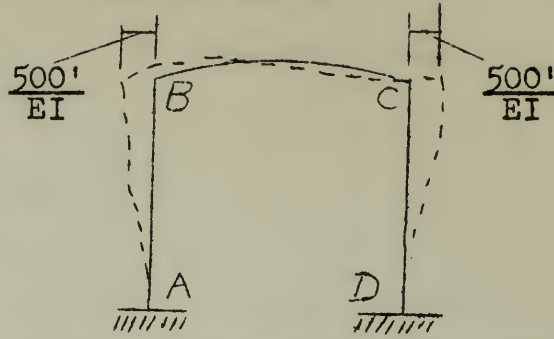
$$x = 7.02$$

$$y = 1.26$$



Thus the AJR_{sp} required to resist spread equals 7.02 kips, and the AJR_{ss} required to resist sidesway equals 1.26 kips.

Assume a spread of $1000/EI$ feet.



For a spread of the arched bent as shown, the following reactions are induced at the ends of the arch:

From Page

$$H_B = 247.6 \frac{EI \Delta_H}{S^3} = \frac{247.6 EI (1000)}{8000 EI} = 30.95 \leftarrow$$

$$H_C = 30.95 \rightarrow$$

$$M_{BC}^F = 432.30 \frac{EI \Delta_H}{S^2} = \frac{32.3 EI 1000}{400 EI} = 480.75$$

$$M_{CB}^F = 480.75$$

with a positive moment indicating tension in the top of the arch. Hence for moment distribution sign convention:

$$M_{BC}^F = 480.75 \text{ ft kips} \quad M_{CB}^F = -80.75 \text{ ft kips.}$$

The moments induced in the ends of the column members are equal to $6 EI \Delta_H / L^2 = \frac{6(500) EI}{EI 400} = 7.5 \text{ ft kips}$

$$\text{and } M_{AB}^F = -7.5 = M_{BA}^F$$

$$M_{CD}^F = 7.5 = M_{DC}^F$$

Solve for moments at the joints caused by the assumed spread.

JOINT	A	B		C		D
MEMB.	AB	BA	BC	CB	CD	DC
COF	+0.500		-0.343		+0.500	
K	0.20	0.20	0.3943	Balancing Moms.		0.20
$\frac{K}{\Sigma K}$	—	0.337	0.663	AT B	AT C	—
FEM	-7.50	-7.5	+80.75			+7.5
	-12.38	-24.75	-48.50	-48.50		
			-12.88	+37.50	+37.50	+9.55
	+2.18	+4.36	+8.52	+8.52	-2.92	
			-0.66	+1.94	+1.94	+0.49
	+0.11	+0.22	+0.44	+0.44	-0.15	+0.02
				+0.10	+0.10	+0.05
M_{sp}	-17.59	-27.67	+27.67			+17.56
BAL. MOMS.			-39.54	+39.54		

The consistent joint force (CJF) required at B and C to maintain the assumed spread is equal to the horizontal restraining forces at the ends of the arch.

HRF caused by:

(a) Spread in the arch

(b) Balancing moment at joint B
.21332(39.54)

(c) Balancing moment at joint C

(d) Shear in member AB

17.59 \neq 27.67

20

(e) Shear in member BC

HRF_L $\xrightarrow{+}$ HRF_R
-30.95^K \neq 30.95^K

\neq 8.50 - 3.50

\neq 8.50 - 8.50

- 2.26

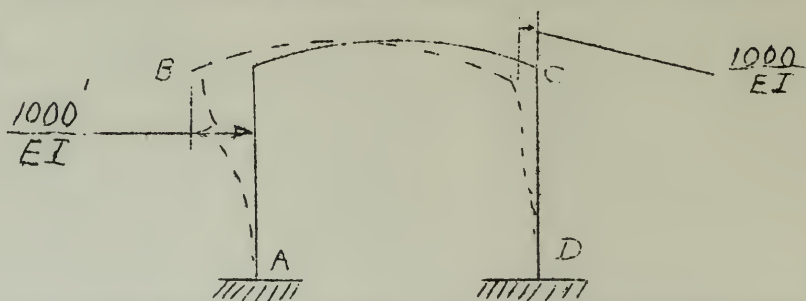
\neq 2.26

CJF_{sp} $\xleftarrow{-16.21^K}$ $\xrightarrow{+16.21^K}$

Thus a consistant joint force of 16.21 kips at each end of the arch acting away from each other will produce the given spread and since an artificial joint restraint of 7.02 kips was required to restrain the original structure from spreading, the actual spread will cause the moments at the various joints to change by an amount equal to

$$\frac{AJR_{sp} (M_{sp})}{CJF_{sp}} = \frac{7.02}{16.21} M_{sp} = 0.432 M_{sp}$$

Next we will assume a sidesway of $1000/EI$ feet to the left



The moments induced in the ends of the column members equals

$$\frac{6EI \Delta_H}{L^2} = \frac{6EI}{400} \frac{1000}{EI} = 15 \text{ foot kips}$$

$$M_{AB}^F = -15 \text{ foot kips}$$

$$M_{BA}^F = -15 \text{ foot kips}$$

$$M_{CD}^F = -15 \text{ foot kips}$$

$$M_{DC}^F = -15 \text{ foot kips}$$

Solve for moments at the joints caused by the assumed sidesway.

JOINT	A			B			C			D		
MEM.	AB	BA	BC				CB	CD	DC			
COF	+0.500			-0.343			+0.500					
K	0.20	0.20	0.3943	Balancing Moment			0.3943	0.200	0.200			
$\frac{K}{\sum K}$	—	0.337	0.663	AT B	AT C		0.663	0.337	—			
FEM	-15.00	-15.00						-15.00	-15.00			
	+2.52	+5.05	+9.95	+9.95			-3.42	+12.22	+6.20	+3.10		
			-4.19		+12.22		-0.96					
	+0.71	+1.41	+2.78	+2.78			+0.63	+0.33	+0.16			
			-0.22									
	+0.03	+0.07	+0.15	+0.15								
M_{ss}	-11.74	-8.47	+8.47				+8.47	-8.47	-11.74			
BAL. MOMS.				+12.88	+12.85							

The consistant joint forces, CJF, required at B and C to maintain the given sidesway are equal to the horizontal restraining forces at C and B and are equal to:

HRF caused by:	HRF _L	HRF _R
(a) Balancing moment at joint B (0.21442)(12.88)	-2.76 ^K	+2.76 ^K
(b) Balancing moment at joint C	+2.76	-2.76
(c) Shear in member AB <u>(11.74)(8.47)</u> 20	-1.01	
(d) Shear in member CD		-1.01
	CJF _{ss} -1.01 ^K	-1.01 ^K

Thus a consistant joint force of 1.01 kips at each end of the arch acting toward the left, will produce the given sidesway, and since an artificial joint restraint of 1.26 kips acting toward the right was required to keep the original structure from swaying, the actual sidesway will cause the moments at the various points to change by an amount equal to $\frac{AJR_{ss}}{CJF_{ss}} M_{ss} = \frac{1.26}{1.01} = 1.248 M_{ss}$

TABULATION OF RESULTS Moments in foot kips.

FACTOR	M _{AB}	M _{BA}	M _{BC}	M _{CB}	M _{CD}	M _{DC}
M _R	+ 1.89	+ 3.77	- 3.77	- 3.13	+ 3.13	+ 1.56
0.432 M _{sp}	- 7.60	-11.92	+11.92	-11.92	+11.92	+ 7.60
1.248 M _{ss}	-14.65	-10.58	+10.58	+10.58	-10.58	-14.65
Final Mom.	-20.36	-18.73	+18.73	- 4.47	+ 4.47	- 5.49
Check by the Conjugate Structure	-20.40	-18.40	+18.40	- 4.70	+ 4.70	-5.80

$$H_A = \frac{(20.36 + 18.73)}{20} = 1.95 \text{ kips} \longrightarrow$$

$$H_D = \frac{(5.49 - 4.47)}{20} = 0.05 \text{ kips} \longrightarrow$$

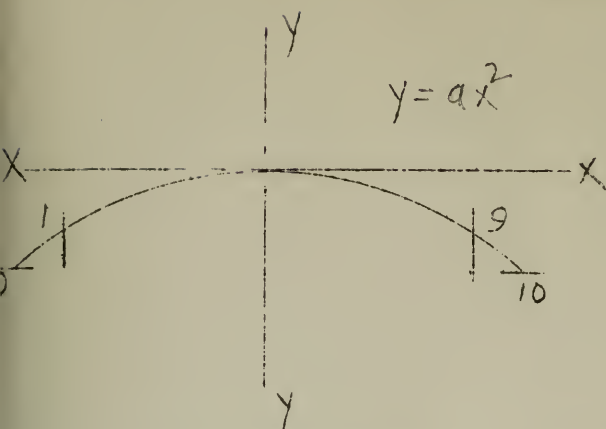
$$V_A = \frac{(40.00 + 60.00 - 20.40 - 5.80)}{20} = 3.69 \text{ kips} \uparrow$$

$$V_D = \frac{(140.00 - 40.00 + 20.40 + 5.80)}{20} = 6.31 \text{ kips} \uparrow$$

APPENDIX A

CALCULATIONS FOR ARC LENGTHS METHOD A

-37-



Infinite series basic equation-

$$(1) \text{ arc lgth } 0-10 = L \left(1 + \frac{2}{3} \left(\frac{2R}{L} \right)^2 - \frac{2}{5} \left(\frac{2R}{L} \right)^4 + \text{etc...} \right)$$

$$(2) \text{ arc lgth } 1-9 = L \left(1 + \frac{2}{3} \text{ etc. } \text{etc. } \text{etc.} \right)$$

Subtract (2) from (1) and divide result by two to get length of segment 0-1

20' Span Rise/Span of 0.2

SEG.	$L=2X$	$R=y$	$\frac{2R}{L}$	$1 + \frac{2}{3} \left(\frac{2R}{L} \right)^2$	$-\frac{2}{5} \left(\frac{2R}{L} \right)^4$	$+ \frac{2}{7} \left(\frac{2R}{L} \right)^6$	Factor
4-5	4	.16	.08	0.0042666	0.0000164	.0000048	1.00425
3-4	8	.64	.16	0.0170666	0.0002621	.000005	1.01681
2-3	12	1.44	.24	0.0383999	0.0013271	.000055	1.03713
1-2	16	2.56	.32	0.0682666	0.0041943	.0003067	1.0643790
0-1	20	4.00	.40	0.1066666	0.0102400	.0011702	1.09760

Total Arc L	$L_a - L_b$	$\frac{L_a - L_b}{2}$	Segment
21.951936			
17.0300640	4.921872	2.460936	0-1
12.4455288	4.584534	2.292267	1-2
8.1344744	4.311054	2.155527	2-3
4.0170008	4.017000	2.058737	3-4
	4.017000	2.00850	4-5

MEMORIAL SERVICE FOR THE LATE

... ..

... ..

...
...
...
...
...
...
...

... ..

...
...
...
...
...

APPENDIX B

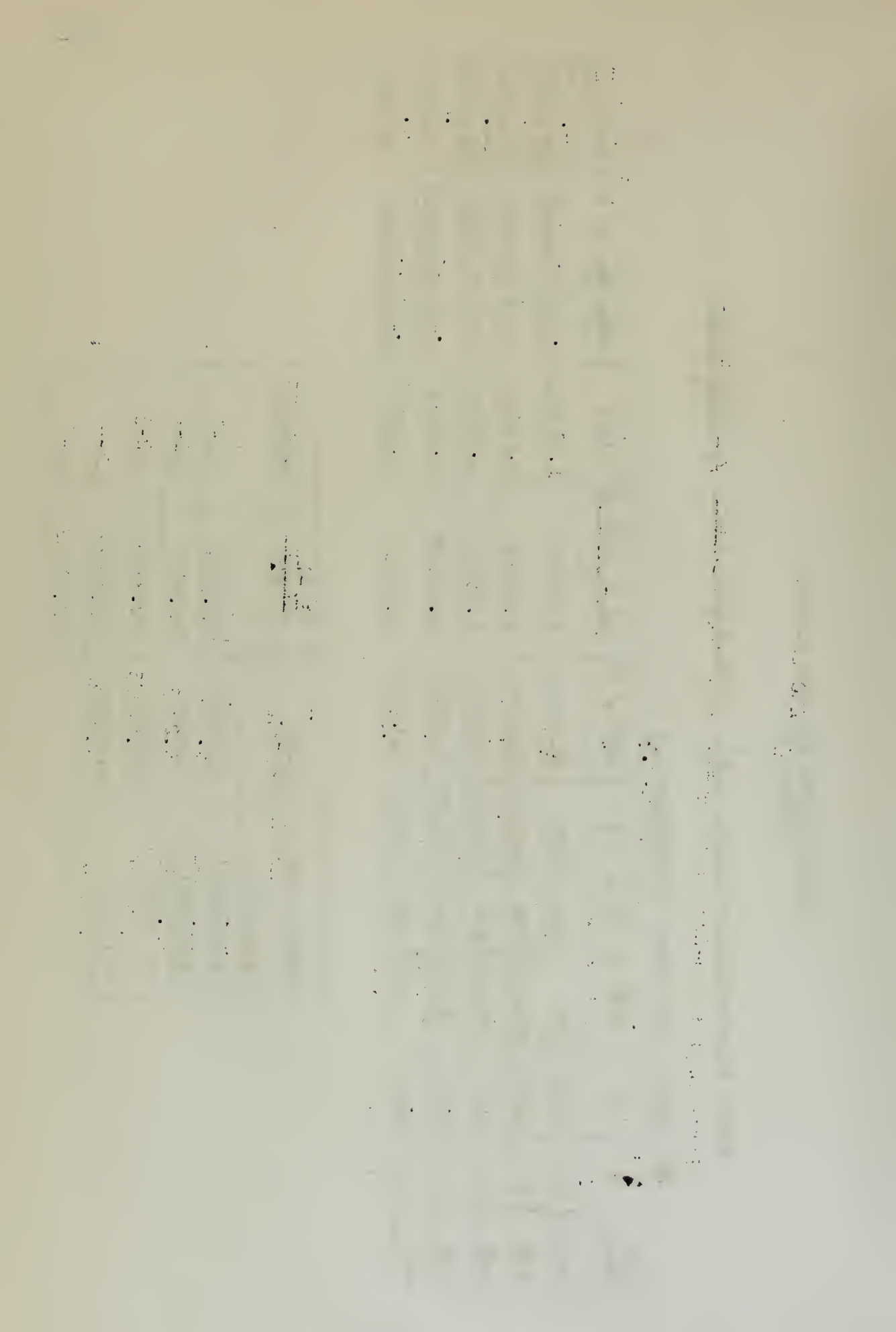
COMPUTATIONS FOR ARC LENGTHS
METHOD B

$$\text{Basic Expression-Arc Length} = \frac{L^2}{8d} \ln \left(\frac{4d + \sqrt{16d^2 + L^2}}{L} \right) + \frac{\sqrt{16d^2 + L^2}}{2}$$

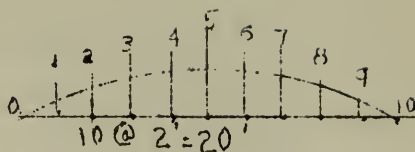
20' Span 6.0' Rise /Span-0.3

ARC SEG.	L	D	16d ² + L ² =	√	16d ² + L ²	$\frac{4d + \sqrt{16d^2 + L^2}}{L}$	ln ()	$\frac{L^2}{8d}$	ln ()	ln() + $\frac{1}{2}$
0-10	20	6.000	576	400	976	31.24099	2.762049	1.015974	8.333	8.46645
1-9	16	3.840	235.929	256	491.92	22.17948	2.34622	0.85280	8.333	7.10666
2-8	12	2.16	74.649	144	218.65	14.78680	1.95223	0.66896	8.333	5.57471
3-7	8	0.96	14.746	64	78.75	8.97388	1.589235	0.46325	8.333	3.860417
4-6	4	0.24	.922	16	16.92	4.11358	1.268395	0.23775	8.333	1.98125
										4.0380

Total Arc L	$\frac{L_a - L_b}{2}$	Segment
24.08695		
18.19641	5.89054	0-1
12.96811	5.22830	1-2
8.29736	4.67075	2-3
4.03804	4.25932	3-4
	2.01902	4-5



TABULATION OF ARC LENGTHS ALL SPANS



Rise/Span-0.04

SEGMENT	20' SPAN	30' SPAN
0-1	2.02066	3.03099
1-2	2.01262	3.01894
2-3	2.00648	3.00972
3-4	2.00240	3.00358
4-5	2.00034	3.00051
0-10	20.0850	30.12750

Rise/Span-0.08

0-1	2.08112	3.121687
1-2	2.050371	3.075556
2-3	2.026073	3.039109
3-4	2.009271	3.013846
4-5	2.001325	3.002048
0-10	20.336320	30.504495

Rise/Span-0.12

0-1	2.178466	3.267699
1-2	2.110352	3.165528
2-3	2.057472	3.086208
3-4	2.2.021372	3.032058
4-5	2.003068	3.004602
0-10	20.741460	31.112190

TABULATION OF ARC LENGTHS (CON'T)

-40-

Rise/SPAN-0.16

SEGMENT	20' SPAN	30' SPAN
0-1	2.304942	3.457414
1-2	2.215587	3.323380
2-3	2.076933	3.115400
3-4	2.037813	3.056720
4-5	2.005447	3.008171
10-0	21.281448	31.922172

Rise/SPAN-0.20

0-1	2.460936	3.691404
1-2	2.292267	3.438402
2-3	2.155527	3.233292
3-4	2.058737	3.088104
4-5	2.008500	3.012750
0-10	21.951936	32.927904

Rise/SPAN-0.30

0-1	2.94527	4.417905
1-2	2.6142	3.9213
2-3	2.33538	3.50307
3-4	2.12966	3.19449
4-5	2.01902	3.02853
0-10	24.08706	36.1306

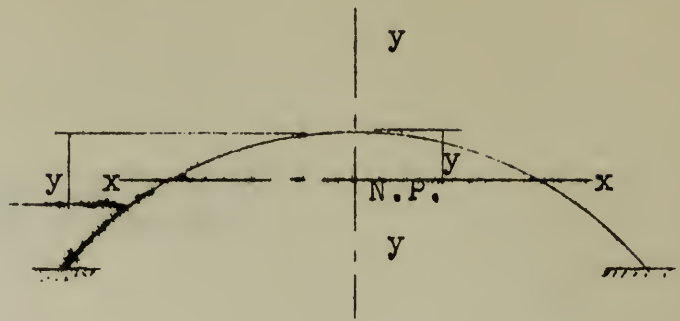
Rise/SPAN-0.40

0-1	3.507935	5.261902
1-2	3.005445	4.508168
2-3	2.565332	3.847998
3-4	2.224695	3.337042
4-5	2.033648	3.050472
0-10	26.674110	40.011166

APPENDIX C

20' Span Rise-0.8
Rise/Span-0.04

$$y = ax^2 \quad z = .008$$



PT.	s = wI	x'	x' ²	y=ax ²	y x s	y'	Sx' ²	sy'	Sy' ²
1	2.0207	-9	81	.648	1.30941	-.3746	163.676	.7569	.2835
2	2.0127	-7	49	.392	.78897	-.1286	98.6223	.2588	.0333
3	2.0065	-5	25	.200	.40130	-.0634	50.1625	.1272	.0081
4	2.0024	-3	9	.072	.14417	-.1914	18.0216	.3833	.0734
5	2.0003	-1	1	.008	.01600	-.2554	2.0003	.5109	.1305
6									
7						I _{yy}	664.9668	I _{xx}	1.0576
8	Lower half of table is same as upper half. All values have been doubled in totals.								
9									
10									

$$\frac{5.31272}{20.0850} = \bar{y} = 0.2649$$

THE UNIVERSITY OF CHICAGO
CHICAGO, ILL.

TO THE PRESIDENT OF THE UNIVERSITY OF CHICAGO
FROM THE FACULTY OF THE DIVISION OF THE PHYSICAL SCIENCES
SUBJECT: A RESOLUTION OF THE FACULTY OF THE DIVISION OF THE PHYSICAL SCIENCES
PASSED AT A MEETING OF THE FACULTY HELD ON MAY 1, 1954

WHEREAS the Faculty of the Division of the Physical Sciences
has considered the report of the Committee on the
Faculty of the Division of the Physical Sciences
and has found it to be satisfactory

IT IS RESOLVED THAT THE FACULTY OF THE DIVISION OF THE PHYSICAL SCIENCES
DOES HEREBY RECOMMEND THAT THE PRESIDENT OF THE UNIVERSITY OF CHICAGO
SHOULD ACCEPT THE REPORT OF THE COMMITTEE ON THE FACULTY OF THE DIVISION OF THE PHYSICAL SCIENCES

TABULATION OF VALUES OF \bar{y}
Centroid Distances

<u>Rise</u> Span	Values of \bar{y}	
	20' Span	30' Span
0.04	0.26490	0.3970
0.08	0.53464	0.80150
0.12	0.81344	1.220146
0.16	1.10528	1.65540
0.20	1.40700	2.11050
0.30	2.22310	3.33465
0.40	3.10472	4.65708

2012

2012		
Jan	1	1
Feb	2	2
Mar	3	3
Apr	4	4
May	5	5
Jun	6	6
Jul	7	7
Aug	8	8
Sep	9	9
Oct	10	10
Nov	11	11
Dec	12	12

CALCULATION OF TRUE MOMENTS OF INERTIA

C.G. / effect of rotation of axes

SECTION	LENGTH	$I_y = \frac{L^3}{12}$	x	$y = ax^2$	$y_a - y_b$	$\theta = \tan^{-1} \frac{y}{x}$	$\frac{\sin^2 \theta}{\cos^2 \theta}$	$I_y \sin^2 \theta$	$I_y \cos^2 \theta$
0-1	2.0207	.6876	8	.512	.288	$8^\circ - 11' - 00''$.0202 .9797	.0139	.6736
1-2	2.0127	.6793	6	.288	.224	$6^\circ - 23' - 00''$.0123 .9876	.0084	.6709
2-3	2.0065	.6732	4	.128	.160	$4^\circ - 34' - 00''$.0063 .9936	.0042	.6689
3-4	2.0024	.6692	2	.032	.096	$2^\circ - 45' - 00''$.0023 .9976	.0015	.6676
4-5	2.0003	.6666	0	.000	.032	$0^\circ - 55' - 00''$.00023 .99976	.00016	.6665

$$\begin{aligned} & @.0281 \quad 6.7020 - I_{yy} \\ & \quad \times 2 \\ & \quad \hline & 0.0562 \end{aligned}$$

These corrections must be added to the I_d^2 values to get actual moments of inertia.

APPENDIX D

0	1	2	3	4	5	6	7	8	9	10	11
//////	ds	M/EI RECTANGLE			M/EI TRIANGLE						
SECT.	LENGTH	HT.	$\frac{d}{dH}$	$\frac{x'}{dV}$	θ 1 x 2	HT.	$\frac{d}{dH}$	$\frac{x'}{dV}$	θ $\frac{1}{2}$ x 1 x 6	RECT. AND TRIANGLE	Δ_v
0-1	3.03099										
1-2	3.01894					#3	-.1360	-10.0	#4.5284	- .6159	- 45.2840
2-3	3.00972	#3	#.0972	- 7.5	9.02916	#3	#.1358	- 7.0	#4.5146	#.8776 # .6131	- 67.7187 - 31.6022
3-4	3.00358	#6	#.2891	- 4.5	18.02148	#3	#.3117	- 4.0	#4.5054	# 5.2100 # 1.4044	- 81.0967 - 18.0216
4-5	3.00051	#9	#.3851	- 1.5	27.00459	#3	#.3917	- 1.0	#4.5008	#10.3995 # 1.7630	- 40.5069 - 4.5008
5-6	3.00051	#12	#.3851	# 1.5	36.00612	#3	#.3757	# 2.0	#4.5008	#13.8659 # 1.6909	# 54.0092 # 9.0016
6-7	3.00358	#15	#.2891	# 4.5	45.0537	#3	#.2637	# 5.0	#4.5054	#13.0250 # 1.1881	#202.7417 # 22.5270
7-8	3.00972	#18	#.0972	# 7.5	54.17496	#3	#.0559	# 8.0	#4.5146	# 5.2658 # .2524	#406.3122 # 36.1168
8-9	3.01894	#21	-.1906	#10.5	63.39774	#3	-.2479	#11.0	#4.5284	-12.0836 - 1.1227	#665.6763 # 49.8124
9-10	3.03099	#24	-.5744	#13.5	72.74376	#3	-.6477	#14.0	#4.5465	-41.7840 - 2.9447	#982.0408 # 63.6510

325.43151

$\theta = +366.0764$

40.644° $\Delta_H = -2.9952$ Δ_v Δ_v

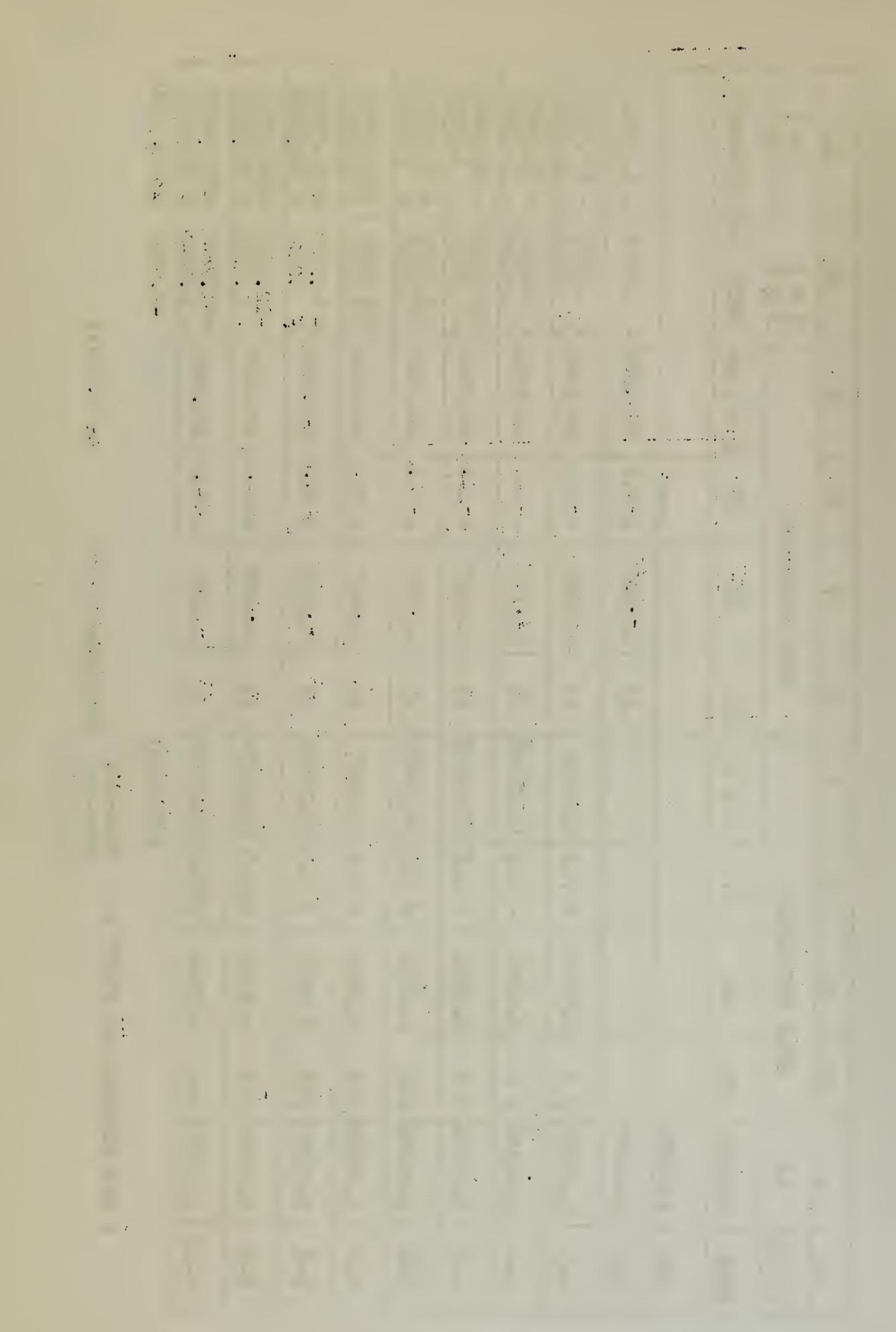
1 KIP VERTICAL AT POINT 1.

Rise-1.2'

Span-30'

y= 0.3970

a= 0.00533



0	1	2	3	4	5	6	7	8	9	10	11
////	ds	M/EI RECTANGLE				M/EI TRIANGLE				5×3 9×7	5×4 9×8
SECT.	LENGTH	HT.	$\frac{y'}{dH}$	$\frac{x'}{dV}$	θ 1 x 2	HT.	$\frac{y'}{dH}$	$\frac{x'}{dV}$	θ $\frac{1}{2} \times 1 \times 6$	RECT. AND TRIANGLE	Δv
0-1											
1-2											
2-3	3.00972					$\neq 3.0$	$\neq 1.1358$	$- 7.0$	$\neq 4.5146$	$\neq .6131$	$- 31.6022$
3-4	3.00358	$\neq 3$	$\neq .2891$	$- 4.5$	$\neq 9.0107$	$\neq 3.0$	$\neq .3117$	$- 4.0$	$\neq 4.5054$	$\neq 2.6049$ $\neq 1.4044$	$- 40.5482$ $- 18.0216$
4-5	3.00051	$\neq 6$	$\neq .3851$	$- 1.5$	$\neq 18.0031$	$\neq 3.0$	$\neq .3917$	$- 1.0$	$\neq 4.5008$	$\neq 6.9329$ $\neq 1.7630$	$- 27.0047$ $- 4.5008$
5-6	3.00051	$\neq 9$	$\neq .3851$	$\neq 1.5$	$\neq 27.0046$	$\neq 3.0$	$\neq .3757$	$\neq 2.0$	$\neq 4.5008$	$\neq 10.3995$ $\neq 1.6906$	$\neq 40.5069$ $\neq 9.0016$
6-7	3.00358	$\neq 12$	$\neq .2891$	$\neq 4.5$	$\neq 36.0430$	$\neq 3.0$	$\neq .2637$	$\neq 5.0$	$\neq 4.5054$	$\neq 10.4200$ $\neq 1.1881$	$\neq 162.1935$ $\neq 22.5270$
7-8	3.00972	$\neq 15$	$\neq .0972$	$\neq 7.5$	$\neq 45.1458$	$\neq 3.0$	$\neq .0559$	$\neq 8.0$	$\neq 4.5146$	$\neq 4.3882$ $\neq .2524$	$\neq 338.5935$ $\neq 36.1168$
8-9	3.01894	$\neq 18$	$- .1906$	$\neq 10.5$	$\neq 54.3409$	$\neq 3.0$	$- .2479$	$\neq 11.0$	$\neq 4.5284$	$- 10.3574$ $- 1.1227$	$\neq 570.5795$ $\neq 49.8124$
9-10	3.03099	$\neq 21$	$- .5744$	$\neq 13.5$	$\neq 63.6508$	$\neq 3.0$	$- .6477$	$\neq 14.0$	$\neq 4.5465$	$- 36.5610$ $- 2.9447$	$\neq 859.2858$ $\neq 63.6510$

30' Span with Rise/Span of 0.04 $\theta = 289.315$ 253.1989

36.1165 -9.3284 ΔH Δv 2030.5905

1 KIP VERTICAL AT POINT 2

$\frac{9.5}{12}$

0	1	2	3	4	5	6	7	8	9	10	11
////	ds	M/EI RECTANGLE			M/EI TRIANGLE						
SECT.	LENGTH	HT.	$\frac{dH}{dV}$	$\frac{dV}{dH}$	$\frac{dH}{dV}$	HT.	$\frac{dH}{dV}$	$\frac{dV}{dH}$	$\frac{1}{2} \times 1 \times 6$	RECT. AND TRIANGLE	$\frac{1}{2} \times 4 \times 8$
0-1											
1-2											
2-3											
3-4	3.00358					$\neq 3$	$\neq 3.1117$	$- 4$	$\neq 4.5054$	$\neq 1.4044$	$- 18.0216$
4-5	3.00051	$\neq 3$	$\neq 3.3851$	$- 1.5$	$\neq 9.0015$	$\neq 3$	$\neq 3.3917$	$- 1$	$\neq 4.5008$	$\neq 3.4665$ $\neq 1.7630$	$- 13.5023$ $- 4.5008$
5-6	3.00051	$\neq 6$	$\neq 3.3851$	$\neq 1.5$	$\neq 18.0031$	$\neq 3$	$\neq 3.3757$	$\neq 2$	$\neq 4.5008$	$\neq 6.9329$ $\neq 1.6909$	$\neq 27.0047$ $\neq 9.0016$
6-7	3.00358	$\neq 9$	$\neq 3.2891$	$\neq 4.5$	$\neq 27.0322$	$\neq 3$	$\neq 3.2637$	$\neq 5$	$\neq 4.5054$	$\neq 7.8150$ $\neq 1.1881$	$\neq 121.6449$ $\neq 22.5270$
7-8	3.00972	$\neq 12$	$\neq 3.0972$	$\neq 7.5$	$\neq 36.1166$	$\neq 3$	$\neq 3.0559$	$\neq 8$	$\neq 4.5146$	$\neq 3.5105$ $\neq .2524$	$\neq 270.8745$ $\neq 36.1168$
8-9	3.01894	$\neq 15$	$- .1906$	$\neq 10.5$	$\neq 45.2841$	$\neq 3$	$- .2479$	$\neq 11$	$\neq 4.5284$	$- 8.6311$ $- 1.1227$	$\neq 475.4831$ $\neq 49.8124$
9-10	3.03099	$\neq 18$	$- .5744$	$\neq 13.5$	$\neq 54.5578$	$\neq 3$	$- .6477$	$\neq 14$	$\neq 4.5465$	$- 31.3380$ $- 2.9447$	$\neq 736.5303$ $\neq 63.6510$

189.9953

$\theta = 221.5972$

$+31.6019$

-16.0128

$\neq 1776.6216$

1 KIP VERTICAL AT POINT 3 30' Span rise/span=0.04

$\frac{I}{S}$

0	1	2	3	4	5	6	7	8	9	10	11
//////	ds	M/EI RECTANGLE				M/EI TRIANGLE				5 x 3 9 x 7	5 x 4 9 x 8
Sect.	LENGTH	HT.	dH	dV	1 x 2	HT.	dH	dV	$\frac{1}{2} \times 1 \times 6$	RECT. AND TRIANGLE	
0-1											
1-2											
2-3											
3-4											
4-5	3.00051					4 3	4.3917	- 1.0	4.5008	4.7630	- 4.5008
5-6	3.00051	4 3	4.3851	4 1.5	4 9.0015	4 3	4.3757	4 2.0	4.5008	4 3.4665 4 1.6909	4 13.5023 4 9.0016
6-7	3.00358	4 6	4.2891	4 4.5	4 18.0215	4 3	4.2637	4 5.0	4.5054	4 5.2100 4 1.1881	4 81.0968 4 22.5270
7-8	3.00972	4 9	4.0972	4 7.5	4 27.0875	4 3	4.0559	4 8.0	4.5146	4 2.6329 4 .2524	4 203.1563 4 36.1168
8-9	3.01894	4 12	- .1906	4 10.5	4 36.2273	4 3	4.2479	4 11.0	4.5284	4 6.9049 - 1.1227	4 380.3867 4 49.8124
9-10	3.03099	4 15	- .5744	4 13.5	4 45.4649	4 3	- .6477	4 14.0	4.5465	- 26.1150 - 2.9447	4 613.7762 4 63.6510

4135.8027 $\Theta = 162.9972$ 27.0965 -20.8835 41468.5263

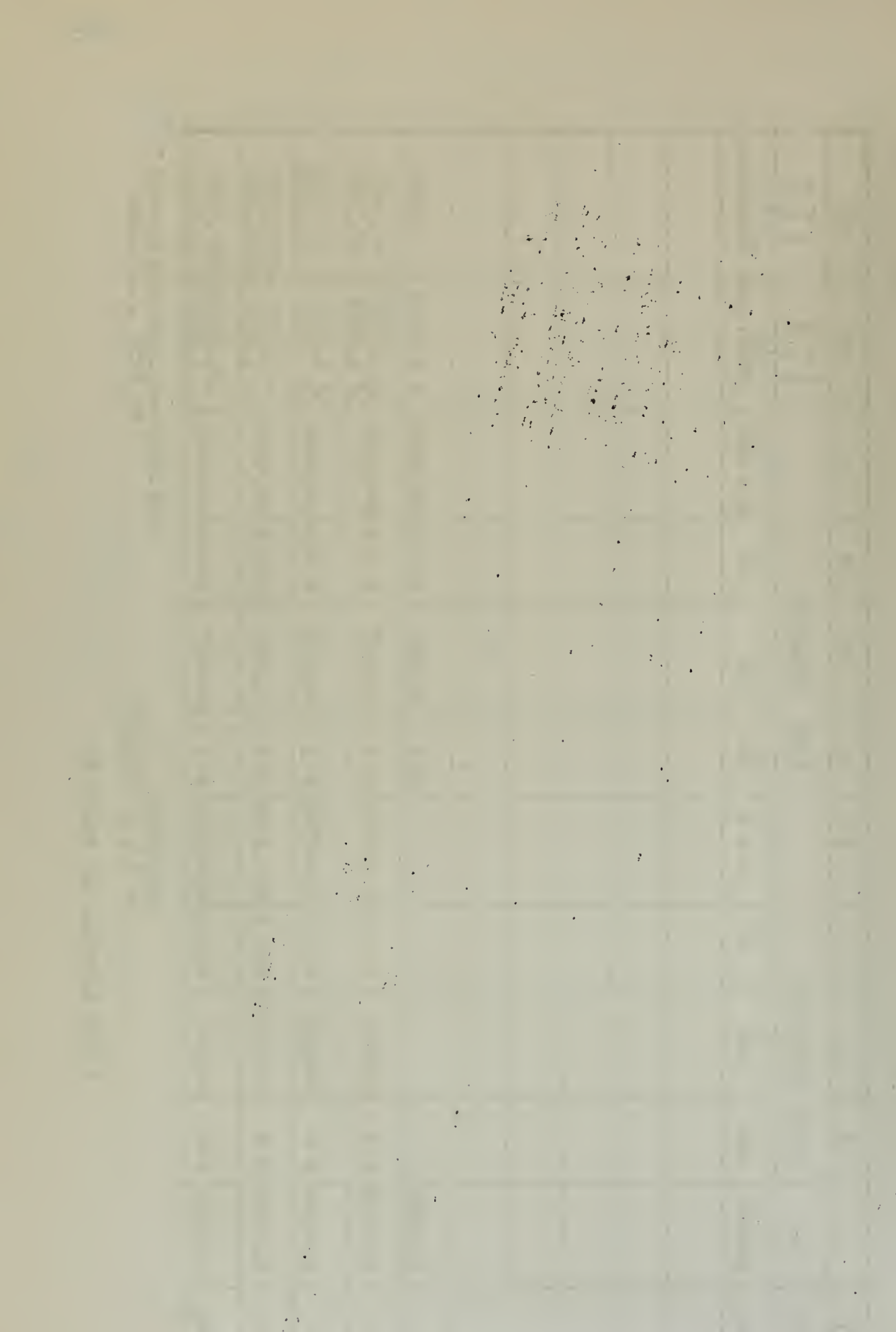
ΔH

1 KIP VERTICAL AT POINT 4. 30' span Rise/Span = 0.04



0	1	2	3	4	5	6	7	8	9	10	11
////	ds	M/EI RECTANGLE		dV		dH		M/EI TRIANGLE		dV	
Sect.	LENGTH	HT.	dH	dV	1 x 2	HT.	dH	dV	1/2 x 1x6	RECT. AND TRIANGLE	Δ_v
5-6	3.00051					4 3	4.3757	4 2.0	44.5008	4 1.6909	4 9.0016
6-7	3.00358	4 3	4.2891	4 4.5	4 9.0107	4 3	4.2637	4 5.0	44.5054	4 2.6050	4 40.5482
7-8	3.00972	4 6	4.0972	4 7.5	418.0583	4 3	4.0559	4 8.0	44.5146	4 1.7553	4135.4373
8-9	3.01894	4 9	4.1906	410.5	427.1705	4 3	4.2479	411.0	44.5284	4 5.1787	4285.2903
9-10	3.03099	412	4.5744	413.5	436.3719	4 3	4.6477	414.0	44.5465	420.8920	4491.0207

$\theta = 5 \neq 9 = 4 113.2071$
 $\neq 90.6114$
1 KIP Vertical at Point #5
 $\neq 22.5957$
 $\neq 21.7104$
 $\neq 113.4053 = \Delta_v$
 $10 \neq 11 = -22.6464 = \Delta_H$
 -0.9360



a = 0.00533

y = 0.3970

Rise- 1.2'

1 KIP Vertical @ N.P.

0	1	2	3	4	5	6	7	8	9	10	11
//////	ds	M/EI	RECTANGLE	$\frac{x'}{dv}$	$\frac{\theta}{1 \times 2}$	HT.	M/EI TRIANGLE	$\frac{x'}{dv}$	$\frac{\theta}{\frac{1}{2} \times 1 \times 6}$	$\frac{5 \times 3}{9 \times 7}$	$\frac{5 \times 4}{9 \times 8}$
Sect.	LENGTH	HT.	$\frac{x'}{dh}$	$\frac{x'}{dv}$	$\frac{\theta}{1 \times 2}$					RECT. AND TRIANGLE	Δ_v
0-1	3.03099	-12	-0.5744	-13.5	-36.3710	-3	-0.6477	-14.0	-4.5465		$\frac{491.0207}{463.6510}$
1-2	3.01894	-9	-0.1906	-10.5	-27.1705	-3	-0.2479	-11.0	-4.5284		$\frac{285.2903}{449.8124}$
2-3	3.00972	-6	0.0972	-7.5	-18.0583	-3	-0.0559	-8.0	-4.5146		$\frac{135.4373}{436.1168}$
3-4	3.00358	-3	0.2891	-4.5	-9.0107	-3	0.2637	-5.0	-4.5054		$\frac{40.5482}{422.5270}$
4-5	3.0005	//////	0.3851	-1.5	//////	-3	0.3757	-2.0	-4.5008		//////
5-6	3.00051	//////	0.3851	1.5	//////	3	0.3757	2.0	4.5008		
6-7	3.00358	3	0.2891	4.5	9.0107	3	0.2637	5.0	4.5054		
7-8	3.00972	6	0.0972	7.5	18.0583	3	0.0559	8.0	4.5146		
8-9	3.01894	9	0.1906	10.5	27.1705	3	-0.2479	11.0	4.5284		
9-10	3.03099	12	-0.5744	13.5	36.3710	3	-0.6477	14.0	4.5465		

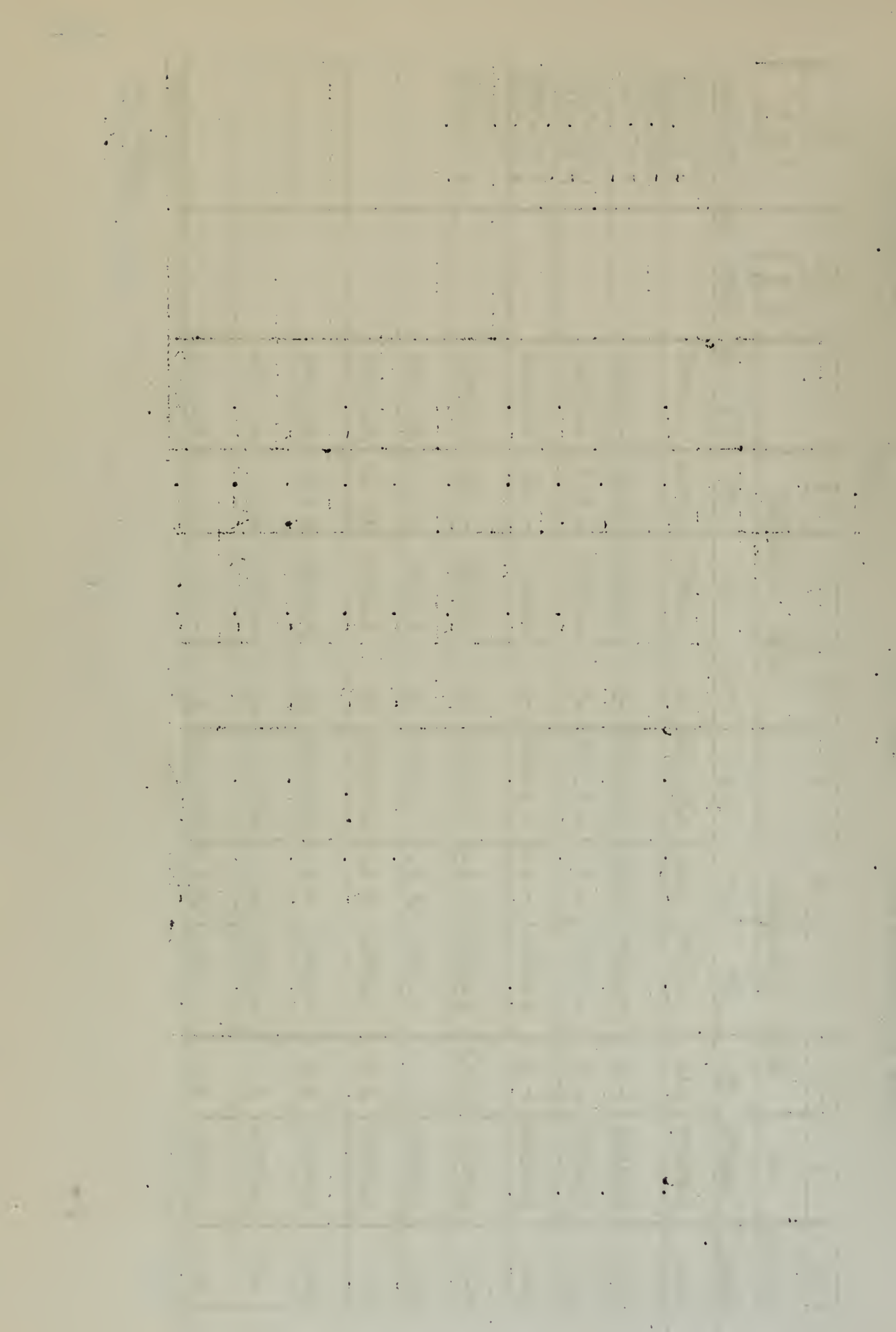
$\sum ds = 30.12750$

0.00

0.00

$\sum \sigma_{uv} = 2266.8106$

$\sigma_{uv} = \frac{1133.4053}{x^2}$



[illegible]

Rise = 1.2'
a = 0.00533
y = 0.3970

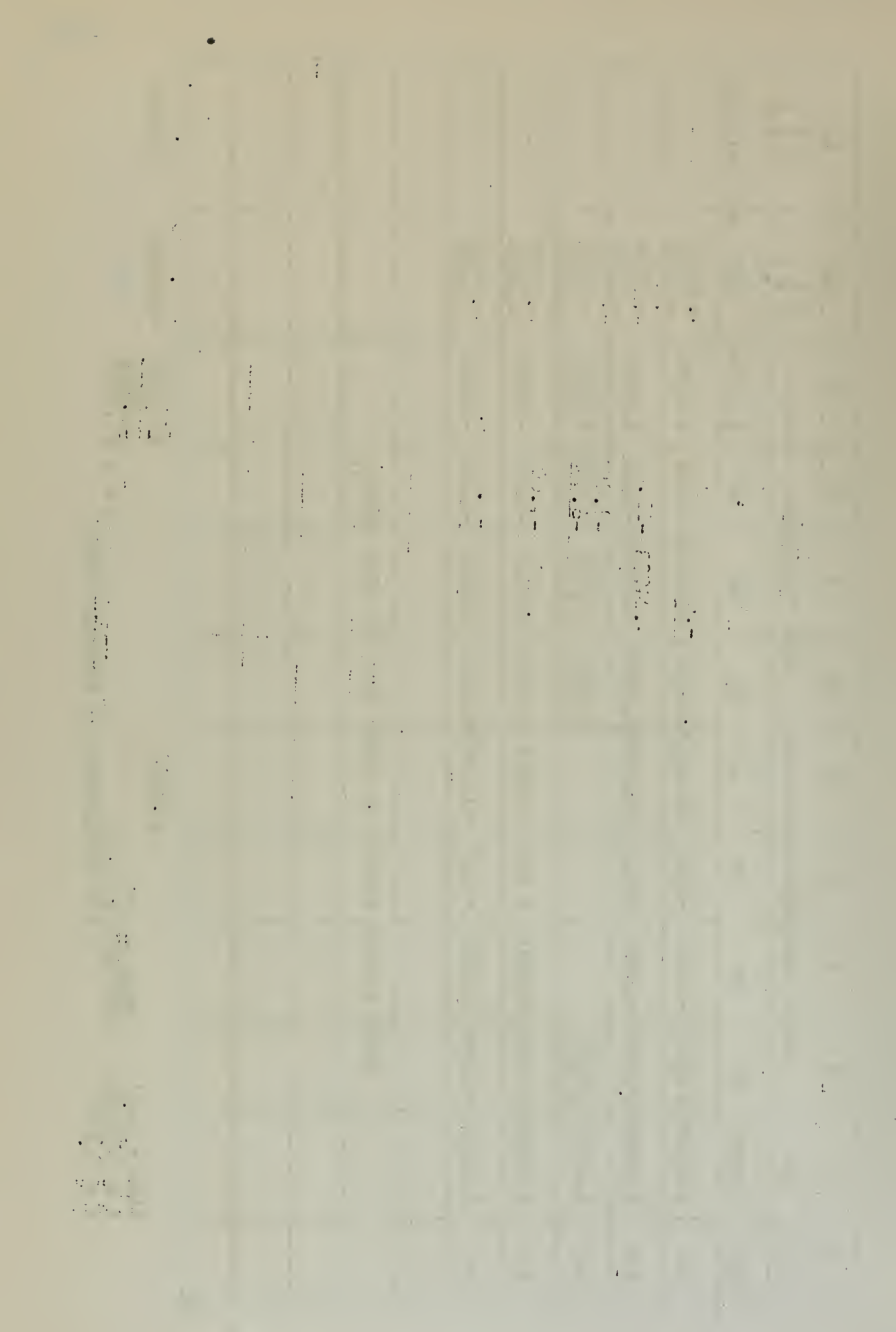
$$\frac{\text{Span} - 30'}{\text{Rise}/\text{Span}} = 0.04$$

4.8712

$$\begin{array}{r} 1.2358 \\ - .8712 \\ \hline = 0.3646 \end{array}$$

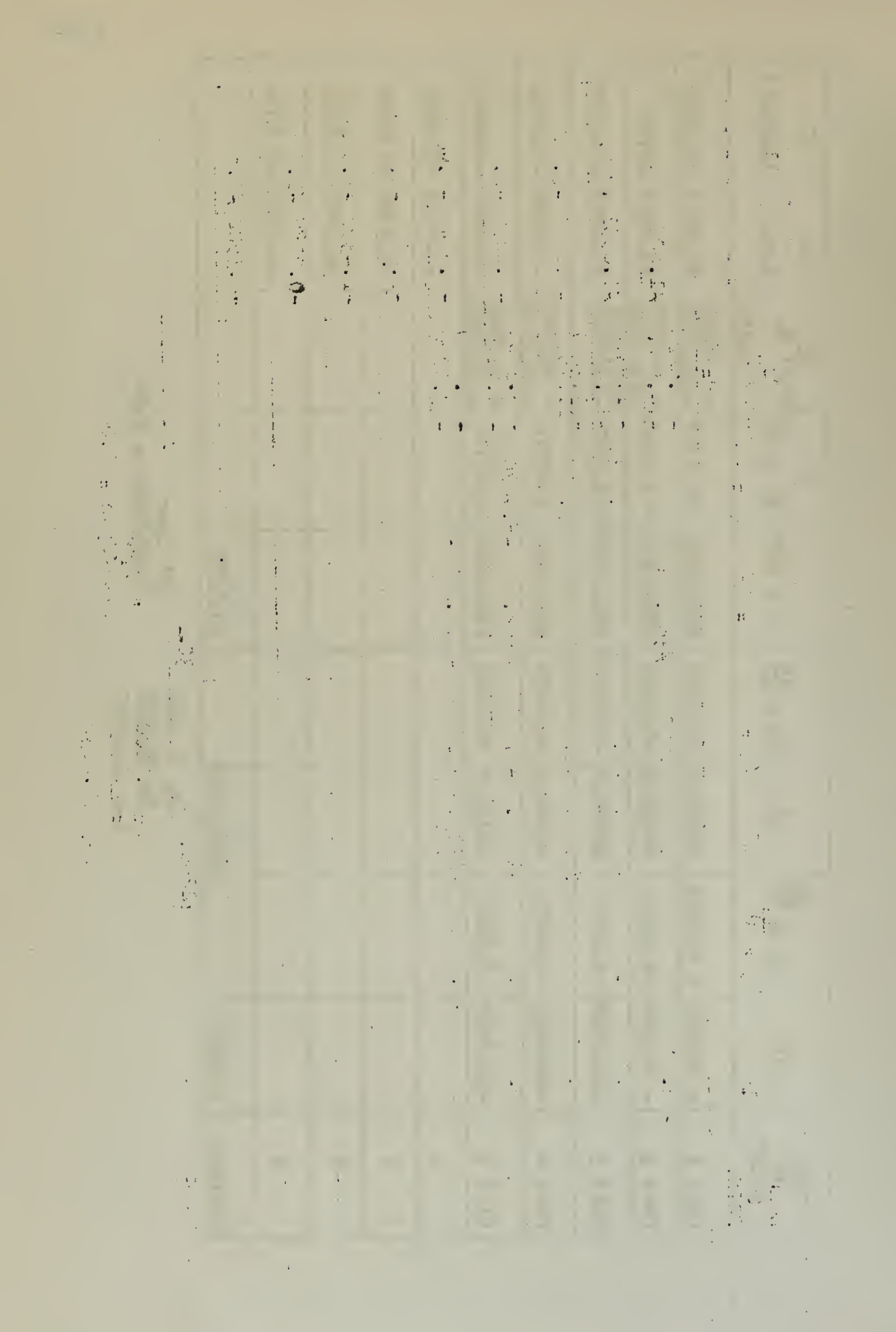
03-8620
NH

5000.0



H_0			V_0	M_0	M_L		M_R	
LOAD @ POINT	10	$H_0 \cdot \frac{dH}{dh}$	11	$V_0 = -\frac{dV}{dV}$	7 $M_0 = \frac{0}{0}$	8 = $(R-y) H_0$ 9 = $15V_0$	10	11
(27) 1	- 0.9952	- 0.7753	2203.1	-0.9719	0 = 549	-12.1509	778-9	778494()
(24) 2	- 9.3284	-2.4145	2030.5	-0.8958	366.076	-14.5788	1.8054	-0.3522
(21) 3	-16.0128	-4.1446	1776.6	-0.78375	289.315	-1.9388 -13.4369	1.8951	9787
(18) 4	-20.8835	-5.4053	1468.5	-0.6478	221.5972	-11.7583	1.0729	-1.4397
(15) 5	-22.6464	-5.8616	1133.4	-0.5000	162.899	-4.3405 -9.7176	-0.0299	-1.4651
6					113.207	-4.7069 -7.5000	-0.9645	-0.9645
7							-1.4651	-0.0299
8							-1.4397	1.0729
9							-0.9787	1.8951
Divisor	d_{HH} 3.8620		d_{VV} 2266.810		d_{em} 30.1275		-0.3522	1.8054

$R = 1.2000$
 $y = 0.3970$
 $R-y = 0.8030$
 $30' \text{ Span}$
 $\text{Rise/Span} = 0.04$



TABULATION OF CONSTANTS BASED ON A UNIT SPAN

-52-

Rise/Span-0.04

Load Point	FEM (LEFT) Ft.Kips	FEM (RT.) ft.Kips	V (left)	H(thrust)
1	/.06036 S	-.01157 S	.9719	.7689
2	/.063325S	-.032465 S	.8958	2.4095
3	/.035915S	-.047835 S	.7838	4.1406
4	-.000855S	-.048695 S	.6478	5.4023
5	-.032025S	-.032025 S	.5000	5.8594

Rise/Span-0.08

1	/.060000 S	-.011805 S	.9717	.3912
2	/.062680 S	-.032550 S	.8952	1.2119
3	/.035370 S	-.047715 S	.7831	2.0741
4	-.001055 S	-.048445 S	.6474	2.7002
5	-.031885 S	-.031885 S	.5000	2.9263

Rise/Span-0.20

1	/.058855 S	-.011685 S	.9705	.1605
2	/.060300 S	-.031840 S	.8921	.4916
3	/.033425 S	-.045975 S	.7794	.8305
4	-.001300 S	-.046260 S	.6449	1.0720
5	-.030335 S	-.030335 S	.5000	1.1581

Rise/Span-0.30

1	/.057900 S	-.011565 S	.9695	.1099
2	/.058525 S	-.030775 S	.8893	.3321
3	/.032060 S	-.043940 S	.7760	.5533
4	-.001162 S	-.043445 S	.6426	.7073
5	-.028210 S	-.028210 S	.5000	.7617

Rise/Span-0.40

1	/.056945 S	-.011655 S	.9686	.0849
2	/.056165 S	-.030935 S	.8871	.2534
3	/.030500 S	-.043205 S	.7732	.4158
4	-.001630 S	-.042330 S	.6407	.5278
5	-.027615 S	-.027615 S	.5000	.5664

APPENDIX E

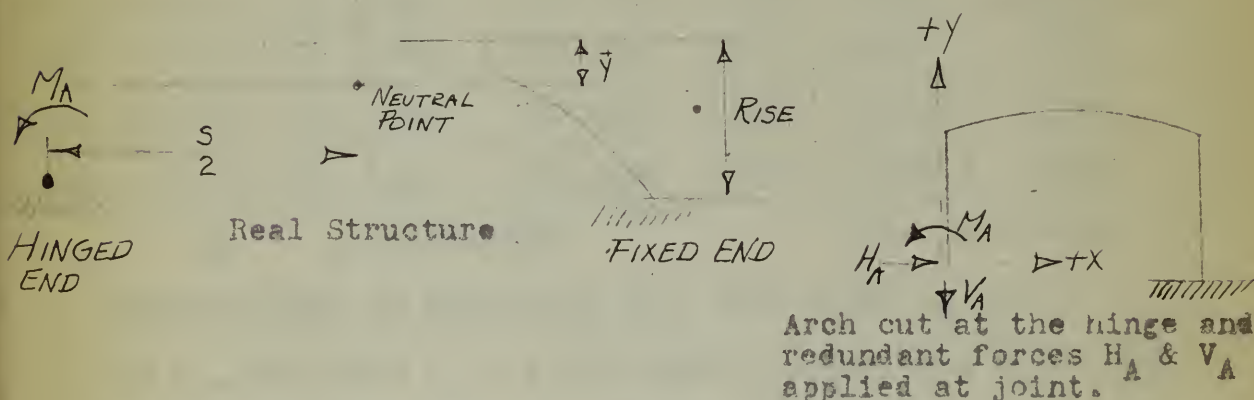
SAMPLE CALCULATIONS FOR DETERMINING:

- (a) Absolute Stiffness
- (b) Carry-over Factor
- (c) Horizontal thrust due to rotation of the joint
- (d) Effect of spread on the end reactions.

Data: 20' Span 0.08' Rise Rise/Span 0.04

Arc Lengths as previously calculated

Neutral point located 0.2649' below crown of arch.



From the theories of virtual work, the following relationships hold true:

$$EI \Delta_V = M_A (EI \delta_{VM}) + V_A (EI \delta_{VV}) + H_A (EI \delta_{VH}) \quad (4)$$

$$EI \Delta_H = M_A (EI \delta_{HM}) + V_A (EI \delta_{HV}) + H_A (EI \delta_{HH}) \quad (5)$$

$$EI \theta = M_A (EI \delta_{\theta M}) + V_A (EI \delta_{\theta V}) + H_A (EI \delta_{\theta H}) \quad (6)$$

FOR OUR SPAN

$$EI \delta_{VM} = EI \delta_{\theta V} = \bar{L} \times \frac{S}{2} = \frac{20.0850(10)}{2} = 200.8500$$

$$EI \delta_{HM} = EI \delta_{\theta H} = \bar{L} (RISE - \bar{y}) = (20.0850)(0.800 - 0.2649) = 10.74748$$

$$EI \delta_{VH} = EI \delta_{\theta V} = \bar{L} (RISE - \bar{y}) \left(\frac{S}{2} \right) = 20.0850 (0.5351) 10 = 107.4748$$

$$EI \delta_{VV} = \sum x^2 ds$$

$$EI \delta_{HH} = \sum y^2 ds$$

ARC	ds in ft	x	x ²	x ² ds	y	y ²	y ² ds
0-1	2.0207	1	1	2.0207	.152	.0231	.0467
1-2	2.0127	3	9	18.1143	.408	.1665	.3350
2-3	2.0065	5	25	50.1625	.600	.3600	.7223
3-4	2.0024	7	49	98.1176	.728	.5300	1.0613
4-5	2.0003	9	81	162.0243	.792	.6273	1.2547
5-6	2.0003	11	121	242.0363	.792	.6273	1.2547
6-7	2.0024	13	169	338.4056	.728	.5300	1.0613
7-8	2.0065	15	225	451.4625	.600	.3600	.7223
8-9	2.0127	17	289	581.6703	.408	.1665	.3350
9-10	2.0207	19	361	729.4727	.152	.0231	.0467

$$EI = 2,673.4868$$

$$EI = 6.8400$$

Substituting in Equations (4), (5), & (6) we get,

$$EI \Delta_H = 200.8500 H_A + 2,673.4868 V_A + 107.4748 H_A \quad (7)$$

$$EI \Delta_V = 10.7475 H_A + 107.4748 V_A + 6.8400 H_A \quad (8)$$

$$EI \theta = 20.8500 M_A + 200.8500 V_A + 10.7475 H_A \quad (9)$$

But since Δ_H & Δ_V are equal to zero, solving equations

$$(7) \text{ and } (8) \text{ simultaneously gives us: } V_A = -0.032474 M_A$$

$$H_A = -1.06101 M_A$$

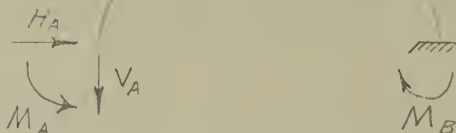
in which H_A is the horizontal thrust due to rotation of the joint and is equal to $-21.22 M_A/S$.

Setting up the free body diagram shown below and solving for M_B

$$M_B = M_A - V_A S$$

$$M_B = M_A - 0.64948 M_A$$

$$M_B = 0.35052 M_A$$



By definition the carry-over factor is equal to the moment induced at the fixed end of a member when a unit moment is applied at the hinged end. Hence the

$$\text{Carry-over factor} = M_B = -0.35052$$

The carry over factor being negative since the induced moment at B causes tension on the same side of the member at the fixed end as the applied moment causes on the hinged end.

Substituting the values of M_A and H_A just obtained in equation (9), the following rotation at joint A occurs;

$$EI\theta = 2.15942 M_A$$

and when θ is equal to one radian

$$M_A = 0.46309 EI \quad (\text{Absolute Stiffness})$$

$$M_A = 9.2618 \frac{EI}{S} \quad (\text{Absolute Stiffness})$$

EFFECT OF SPREAD

When spread occurs with no rotation or vertical deflection at the ends of the arch, equations (7) and (9) become equal to zero. Solving them simultaneously,

$$V_A = 0$$

$$H_A = -1.86881 M_A$$

$$M_A = -0.53510 H_A$$

Substituting in equation (8) and solving for H_A and M_A in terms of $EI \Delta_H$

$$M_A = -0.491328 EI \Delta_H$$

$$H_A = 70.918206 EI \Delta_H$$

Solving several span lengths, with the same rise/span ratio, it was determined that M_A varied inversely as the square of the span, and H_A varied inversely as the cube of the span. Therefore,

$$M_A = -196.53 \frac{EI \Delta_H}{S^2}$$

$$H_A = 7,345.6 \frac{EI \Delta_H}{S^3}$$

rise/span	absolute stiffness	carryover
0.04	9.2618 EI/S	-0.3505
0.08	9.0522 EI/S	-0.3497
0.20	7.8855 EI/S	-0.3431
0.30	6.7742 EI/S	-0.3384
0.40	5.8205 EI/S	-0.3362

When joint A is free to rotate, and a moment of M_a ft. kips is applied at A, the following thrusts and shears are induced at A.

rise/span	thrust kips	vertical shear kips
0.04	21.220 EI/S M_a	0.6495EI/S M_a
0.08	10.629 EI/S M_a	0.6503EI/S M_a
0.20	4.288 EI/S M_a	0.6569EI/S M_a
0.30	2.894 EI/S M_a	0.6616EI/S M_a
0.40	2.203 EI/S M_a	0.6638EI/S M_a

When a spread of Δ_H feet occurs, the following moments, shears, and thrust occur:

rise/span	Thrust	Moment	Shear
0.04	7346	-196.5312	0.00
0.08	1806.3	-96.2225	0.00
0.20	247.64	-32.3053	0.00
0.30	103.83	-19.6084	0.00
0.40	52.391	-12.8234	0.00

NOTE: recorded values of thrust must be multiplied by $\frac{EI \Delta_H}{s^3}$

and all moments by $\frac{EI \Delta_H}{s^2}$

BIBLIOGRAPHY

1. "INDETERMINATE STRUCTURES" - Prof. J. S. Kinney
2. "THEORY OF MODERN STEEL STRUCTURES" -Vol. II Grinter
3. "THEORY AND PRACTICE OF REINFORCED CONCRETE" -C. W. Dunham
4. "CONCRETE INFORMATION", Pamphlet ST41, issued by the Structural Bureau of the Portland Cement Association.
5. "APPLICATION OF MOMENT DISTRIBUTION TO ARCHED BENTS"
Masters thesis of M. G. Cain Jr. RPI 1947.

DATE DUE

[illegible]

Thesis
H2

6885

Hansen

Method of application
of moment distribution
to solution of arched
bents.

Thesis
H2

6885

Hansen

Method of application
of moment distribution
to solution of arched
bents.

thesH2

Method of application of moment distribu



3 2768 002 07639 0

DUDLEY KNOX LIBRARY